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# THESIS

SPECIFICATION OF MARINE SURFACE

BOUNDARY LAYER PARAMETERS

by

Donald Reed McConathy, Jr.

Thesis Advisor:

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March 1972

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Iteration criteria used in the model were evaluated with respect to operational application. Winds calculated by the model at the height of 19.5 meters were incorporated into the current FNWC wave height program, and the results compared with observed wave data. Turbulent parameters describing the properties of the index of refraction ( $C_n^2$ ) were also computed and evaluated with respect to laser propagation criteria.

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Specification of Marine Surface  
Boundary Layer Parameters

by

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$a_h$	Reciprocal of the turbulent Prandtl number ( $K_H/K_m$ )
$B$	Boston constant = 0.81
$B_o$	Blackadar constant = 0.0003
$C_n^2$	Refractive index structure constant
$C_p$	Specific heat of air at constant pressure
$C_T^2$	Structure constant for temperature fluctuations
$C_z$	Nondimensional drag coefficient at level $z$
$\epsilon$	Rate of dissipation of turbulent energy
$\eta$	Angle between the geostrophic wind and the thermal wind
$f$	Coriolis parameter
$G$	Surface geostrophic wind speed
$g$	Gravity = 9.8 meters/second
$H$	Upward turbulent heat flux
$h$	Height of surface layer
$K$	Streamwise component of wave number
$K_c$	Curvature
$K_H$	Turbulent transfer coefficient for heat
$K_m$	Turbulent transfer coefficient for momentum
$k$	von Kármán constant
$L$	Lettau - Monin - Obukhov stability length
$L'$	Modified stability length

$L_*$	Dimensionless stability length
$N$	Measure of the "smearing" of temperature inhomogeneities
$\nu$	Kinematic viscosity
$p, q, s$	Baroclinic model parameters
$p'$	Dimensionless baroclinic model parameter
$P$	Surface pressure (millibars)
$\theta_D$	Divergence terms
$\theta_t$	Nondimensional temperature gradient
$\theta_\theta$	One-dimensional spectrum for temperature fluctuation
$\theta_u$	Nondimensional wind shear
$\Psi$	Integrated profile stability parameter
$\Psi_0$	Surface geostrophic wind angle
$R$	Radius of curvature
$R_d$	Gas constant for dry air
$R_i$	Richardson number
$\bar{\rho}_a$	Mean density of surface boundary layer
$r$	Dimensionless thermal wind
$T_a$	Surface air temperature ( $^{\circ}\text{C}$ )
$T_s$	Sea surface temperature ( $^{\circ}\text{C}$ )
$\theta$	Potential temperature ( $^{\circ}\text{K}$ )
$\theta_*$	Scaling temperature
$\theta_a$	Air temperature measured at level of thermometer ( $^{\circ}\text{K}$ )

$\theta_o$	Potential temperature at the surface ( $^{\circ}$ K)
$\theta_s$	Sea surface temperature ( $^{\circ}$ K)
$\tau$	Surface stress
$U_*$	Friction velocity
$U_h$	Wind at level $h$
$U_o$	Surface wind
$U_z$	Wind at height $z$
$u_g, v_g$	$u, v$ components of geostrophic wind
$u_T, v_T$	$u, v$ components of thermal wind
$v_G$	Gradient wind
$W$	Ageostrophic wind component
$W_A$	Wave height (feet)
$z$	Height above sea level (meters)
$z_o$	Surface roughness parameter (meters)

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## I. INTRODUCTION

Parameterization of the boundary layer over the sea surface has been receiving increased attention in recent years. Fortunately much of the theory developed for low level turbulence over a land surface is also applicable over the oceans. But the air-water interface does present some unique problems. For instance a fluid surface, rather than a fixed one, results in mobile surface roughness elements. Because of the presence of a fluid surface, any parameterization requires a knowledge of the processes which transfer heat and momentum energy across the boundary and ultimately influence such variables as wave generation and index of refraction.

The purpose of this study was to examine and evaluate several features of a recently formulated marine boundary layer model which was designed for operational use. The model is a two-layer baroclinic model for the marine boundary layer and the formulation examined was that developed by Cardone (1969). Cardone based his formulation on an earlier two-layer model due to Blackadar (1965a). Features of interest in the model are related to the physics of the planetary boundary layer and include the effect of atmospheric stability, baroclinicity, coriolis acceleration and changes in the surface roughness parameter on the pressure driven surface wind.

The scope of the study was to examine the model in the context of its operational usage. As such, it was a case study in which the data represented pressure and temperature observations from sequential 12-hour periods when a large cyclonic pressure system dominated the Central North Pacific Ocean. This particular synoptic situation provided the necessary relations between the scalar fields to examine features associated with baroclinicity, stability and wind speed categories.

Because these data represent scalar fields which had to be defined by some analysis scheme before being used as input to the two-layer model, the differences which arise due to procedures in the external field specifications are considered. The effect of isobaric curvature on the surface wind fields is examined and this effect is compared to the changes associated with the physics in the model. The differences which may arise solely because the fields are defined by an objective versus a subjective analysis are considered.

Finally, expressions describing the boundary layer are, by their very nature, empirical. As such, they contain empirical constants and the von Kármán constant ( $k$ ) is one whose value has recently been questioned. The importance of a suggested change from  $k=0.40$  to  $0.35$  was considered.

## II. ASPECTS OF A PLANETARY BOUNDARY LAYER

### A. SURFACE BOUNDARY LAYER

#### 1. Basic Structure of the Boundary Layer

Recent advances in the study of atmospheric turbulence have led to a greater confidence in describing the distribution of wind and temperature in the atmospheric boundary layer over a solid surface. This progress has arisen largely through the application of the Monin-Obukhov similarity theory which has been supported by recent observational studies. Cardone's contribution was primarily to extend Blackadar's two-layer neutral baroclinic boundary layer model over fixed terrain, and then to apply this model to the boundary layer over a fluid surface. The major difference between these boundaries is that the sea surface changes its character as the wind blows over it.

In general, Blackadar's model treats the boundary layer as being formed by both turbulent friction and the coriolis acceleration. All processes are considered to be three dimensional, and the layer is characterized by a density stratification. It is conveniently separated into the three regions depicted in Figure 1. In this delineation the important consideration is the specification of  $K_m$  which is the turbulent transfer coefficient for momentum.

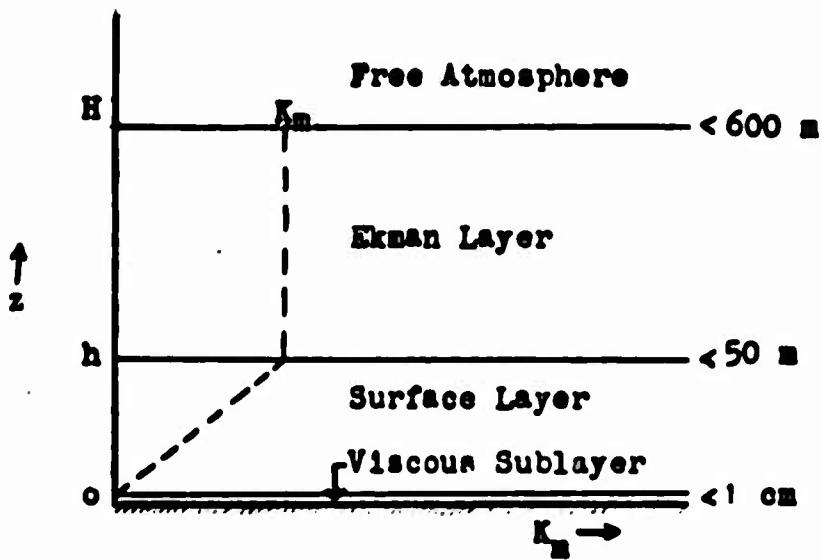


Fig. 1. Diagram of the planetary boundary layer

a. Viscous Sublayer (less than 1 cm)

In this near surface layer the stress is supported by viscosity (aerodynamically smooth flow), and the mean velocity increases linearly according to

$$\bar{U}_z = \frac{U_*^2 z}{v} \quad (1)$$

where  $v$  is the kinematic viscosity and  $U_*$  is the friction velocity.

b. Surface Layer (up to 50 meters)

This is the constant stress layer in which the coriolis parameter can be neglected and  $K_m$  increases linearly with height. For neutral stratification the logarithmic law is valid

$$\bar{U}_z = \frac{U_*}{k} \log z/z_0 \quad (2)$$

where  $k=0.4$  is the von Kármán constant.

### c. Ekman Layer (50-600 meters)

This layer is characterized by the stress decreasing with height and  $K_m$  constant throughout the layer.

In formulating the boundary layer model, Cardone (1969) considered the validity of the "log law" for the surface layer over water, which has been experimentally documented down to near surface limits (Phillips, 1966; Paulson, 1967). He also assumed that the process of wave generation did not seriously affect the surface layer since under neutral and steady conditions a logarithmic shear zone should be established. Accepting the log profile as being valid, Cardone was able to apply much of the surface layer turbulence theory developed over land to over-water conditions. However Cardone did take into account the action of the waves when specifying the effective roughness parameter for the sea surface.

## 2. Similarity Theory for the Surface Layer

The Monin-Obukhov "similarity theory" has led to considerable progress in the specification of the distribution of wind and temperature in the boundary layer. The basis of this theory is that near the ground there exist velocity ( $U_*$ ), length ( $L$ ), and temperature ( $\theta_*$ ) scales which are essentially invariant with height. When the principle variables such as temperature ( $T$ ), wind ( $U$ ), and height ( $z$ ) are expressed

as nondimensional fractions of these quantities, a series of nondimensional equations can be formed which are of general validity in the surface boundary layer (Lumley and Panofsky, 1964). Recent analyses of the oceanic wind data (e.g. Phelps and Pond, 1971) have indicated that the Monin-Obukhov theory is valid for over-water application. These results support Cardone's inclusion of the Monin-Obukhov stability criteria in the boundary layer model.

Similarity theory predicts that a universal relation should exist for the nondimensional wind shear

$$\frac{kz}{U_*} \frac{\partial U}{\partial z} = \beta_u \quad (3)$$

and the nondimensional temperature gradient

$$\frac{z}{\theta_*} \frac{\partial \theta}{\partial z} = \beta_t \quad (4)$$

where  $\beta_t$  and  $\beta_u$  are functions of the dimensionless ratio  $z/L$ . Here  $\theta$  is the potential temperature,  $\theta_*$  is the scaling temperature defined by

$$\theta_* = - \frac{1}{kU_*} \frac{H}{C_p \rho_a} \quad (5)$$

and  $L$  is the Lettau-Monin-Obukhov stability length defined by

$$L = \frac{-U_*^3 C_p \bar{\rho}_a T}{k g H} \quad (6)$$

The heat flux ( $H$ ) is not easily measured, so a modified stability length was defined by Panofsky (1963)

$$L' = a_h L = \frac{K_H}{K_m} L = \frac{U_* T \partial u / \partial z}{k g \partial \theta / \partial z} \quad (7)$$

The assumption that  $L'$  is independent of height implies that  $a_h$  is constant, which in turn implies similarity between wind and temperature profiles.

If  $L'$  is a valid scaling length, similarity theory predicts that

$$\frac{kz}{U_*} \frac{\partial U}{\partial z} = \beta_u(z/L') \quad (8)$$

$$\frac{z}{\theta_*} \frac{\partial \theta}{\partial z} = \frac{\beta_u(z/L')}{a_h}. \quad (9)$$

Integration of these formulae (Panofsky, 1963) yields

$$U_z = \frac{U_*}{k} \left[ \ln z/z_0 - \Psi(z/L') \right] \quad (10)$$

$$\theta_z = \theta_0 + \frac{\theta_*}{a_h} \left[ \ln z/z_0 - \Psi(z/L') \right], \quad (11)$$

where the relation between  $\Psi$  and  $\beta_u$  is

$$\Psi(z/L') = \int_0^{z/L'} \frac{1 - \beta_u(\xi)}{\xi} d\xi, \quad \xi = z/L'. \quad (12)$$

Paulson (1970), using the KEVPS formula, integrated (12) by parts to obtain the following expression for  $\Psi$

$$\Psi = 1 - \beta_u - 3 \ln \beta_u + 2 \ln \left( \frac{1 + \beta_u}{2} \right) + 2 \tan^{-1} \beta_u - \frac{\pi}{2} + \ln \left( \frac{1 + \beta_u^2}{2} \right). \quad (13)$$

Within the scope of an operational application, these expressions for the surface layer would be used to determine the wind speed given  $U_*$ , or to determine  $U_*$  given the wind speed at a particular level. For this reason the partial derivatives in (7) are replaced by expressions involving height differences and air-sea temperature

differences. This is achieved by substituting the following expression for  $\frac{\partial \theta}{\partial z}$ , obtained by combining (9) and (11)

$$\frac{\partial \theta}{\partial z} = \frac{\theta_u \left( \frac{z_a}{L'} \right) (\theta_a - \theta_s)}{z \left[ \ln \left( \frac{z_a}{z_0} \right) - \Psi \left( \frac{z_a}{L'} \right) \right]}, \quad (14)$$

where  $z_a$  is the height at which  $\theta_a$  is measured (chosen as 10 meters by Cardone).

Substituting (14) and (8) into (7) yields

$$L' = \frac{U_*^2 \bar{\theta} \left[ \ln \left( \frac{z_a}{z_0} \right) - \Psi \left( \frac{z_a}{L'} \right) \right]}{k^2 g (\theta_a - \theta_s)}. \quad (15)$$

Equation (10) is then solved for  $U_*$

$$U_* = \frac{k U_m}{\ln \left( \frac{z_m}{z_0} \right) - \Psi \left( \frac{z_m}{L'} \right)}, \quad (16)$$

where  $z_m$  is the height at which the wind speed is measured (chosen as 19.5 meters by Cardone). Equations (15) and (16) can then be solved for  $U_*$  given the wind speed at some level ( $U_z$ ) and the air-sea temperature difference.

### 3. Consideration of the Drag Coefficient ( $C_d$ ) and Roughness Parameter ( $z_0$ )

The surface roughness parameter and the influence of thermal stratification have received considerable attention in the definition of the relation between the nondimensional drag coefficient ( $C_d$ ) and the wind speed at a certain height (usually 10 meters). The need for such a relation arises from the "bulk aerodynamic" formula for estimating

momentum flux. This equation, deduced from dimensional analysis, has the form

$$\tau = \bar{\rho}_a U_*^2 = \bar{\rho}_a C_z U_z^2 , \quad (17)$$

where  $\tau$  is the surface stress. From this expression the drag coefficient is defined as

$$C_z = \frac{U_*^2}{U_z^2} . \quad (18)$$

Rearranging and squaring (10) yields the following expression for the drag coefficient at a level  $z$  in terms of  $z_o$  and stability

$$C_z = \left[ \frac{k}{\ln z/z_o - \Psi(z/L')} \right]^2 . \quad (19)$$

which was the form considered by Cardone.

Observations have indicated that over the sea,  $z_o$  is dependent on  $U_*$ . This dependence, however, changes as the wind speed changes. For the case of light winds (aerodynamically smooth flow), the drag coefficient  $C_z$  decreases with increasing wind speed (Kraus, 1966) and  $z_o$  has the form

$$z_o = \frac{V}{1.09 U_*} . \quad (20)$$

For aerodynamically rough flow,  $C_z$  increases with increasing wind speed according to

$$z_o = \frac{a U_*^2}{g} \quad (21)$$

where (a) is a constant = 0.035 (Charnock, 1955). A summary of field

observations indicates that the change between rough and smooth flow occurs at wind speeds of 5-6 meters/second.

Cardone proposed the following simple expression for  $z_c$  covering a wide range of wind speeds (both rough and smooth flow)

$$z_o = \frac{C_1}{U_*} + C_2 U_*^2 + C_3. \quad (22)$$

The constants were chosen so that  $C_z$  was a minimum for a value of 6 meters/second at the 10 meter height under neutral conditions, and above this speed the relation approximated Charnock's results. For  $U_*$  in meters/second and  $z_o$  in meters, the resulting expression was

$$z_o = 6.84 \times 10^{-5}/U_* + 4.28 \times 10^{-3} U_*^2 - 4.43 \times 10^{-4}. \quad (23)$$

Other representations of  $C_z$  versus  $U_z$  have been published. A few of these appear in Figure 2 along with Cardone's approximation. It is clear from the scatter of curves in Figure 2 that the specification of the drag coefficient was a primary decision in Cardone's formulation. This choice of  $C_z$  versus  $U_z$  will be examined during the discussion on results from this investigation.

With the equations for stability length (15), friction velocity (16), and surface roughness parameter (23), the surface layer wind distribution is completely described within the framework of the similarity theory. These three equations can be solved simultaneously for  $U_*$  using iteration techniques with the external parameters described below as input.

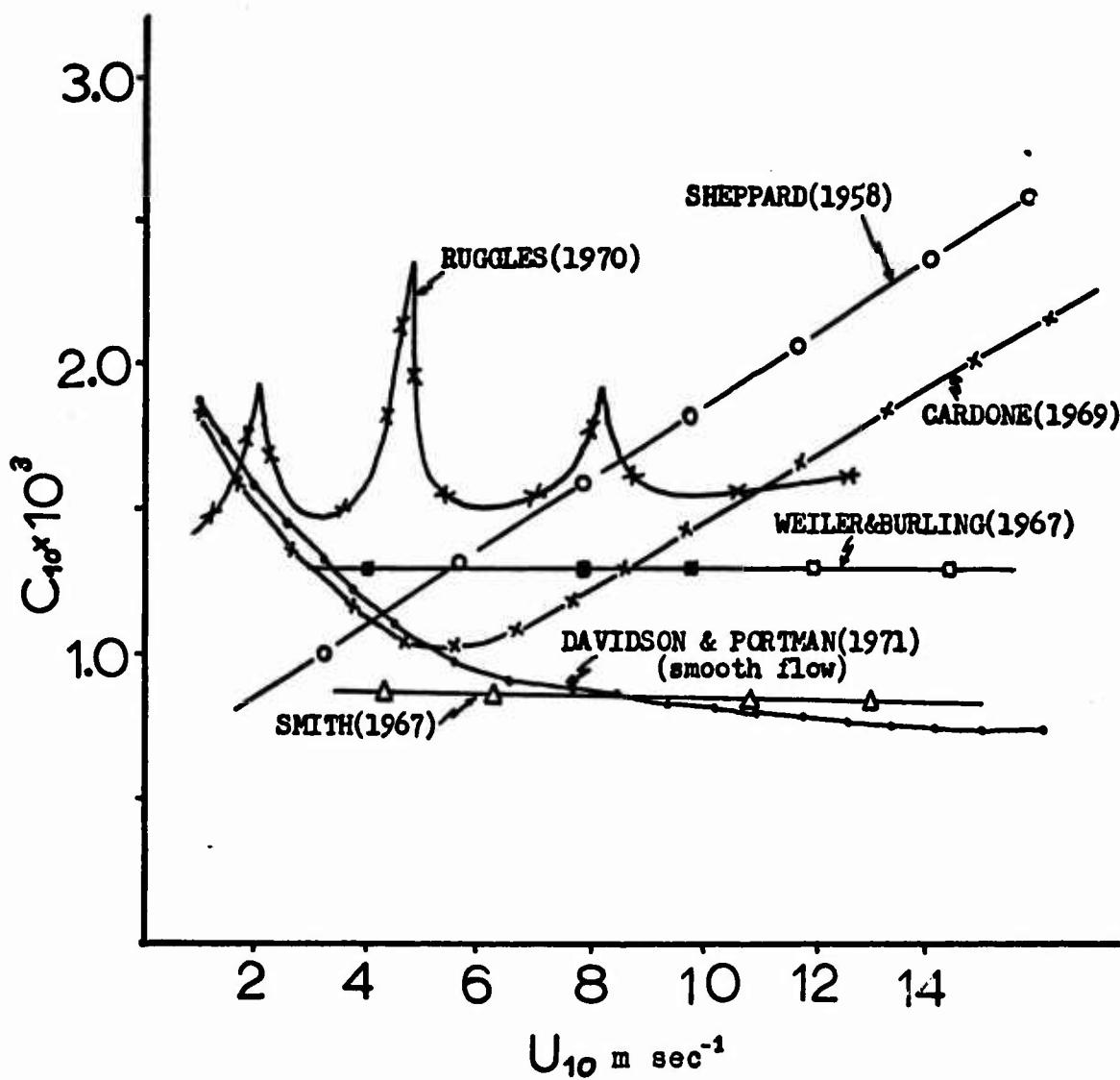


Fig. 2. Various suggested forms of the drag coefficient ( $C_d$ ) versus windspeed ( $U_z$ ) at 10 meters

## B. PLANETARY BOUNDARY LAYER

### 1. Background

In developing a model for the entire boundary layer, consideration has to be given to the possible external parameters. In Cardone's formulation these parameters were the geostrophic wind ( $G$ ), the air-sea temperature difference ( $\theta_a - \theta_s$ ), the coriolis parameter ( $f$ ), and the thermal wind ( $r$ ). He considered previously suggested empirical relations between the geostrophic drag coefficient ( $U_*/G$ ) and the surface Rossby number (Lettau, 1959)

$$R_o = \frac{G}{f z_o} \quad (24)$$

where  $G$  is the surface geostrophic wind speed. Empirical relations have also been suggested between the cross isobar flow angle  $\Psi$  and  $R_o$ .

Blackadar (1965a) developed a two-layer boundary layer model, consisting of a Prandtl surface layer and an Ekman layer, which was verified on the basis of the above relations.

Cardone's formulation was closely related to Blackadar's work. In particular, the height of the surface layer ( $h$ ) was specified explicitly in terms of the external parameters

$$h = \frac{B_o G}{f} \quad (25)$$

where  $B_o$  was a nondimensional constant equal to  $3.0 \times 10^{-4}$  as determined by Blackadar.

By applying proper boundary conditions and requiring that continuity of wind and wind shear exist at level  $h$ , the following solutions for  $U_*/G$  and  $\Psi_o$  (along with the solution for  $z_o$ ) completely define the surface boundary layer wind distribution under neutral conditions.

$$\frac{U_*}{G} = \left[ 2 k B_o \sin^2 \Psi_o \right]^{3/2} \quad (26)$$

$$\frac{U_*}{G} = \frac{k \sqrt{2} \sin (\pi/4 - \Psi_o)}{\ln B_o R_o} \quad (27)$$

Implicit in such an approach is that the value of  $K_m$  at the internal boundary between the surface layer and the Ekman layer is given by

$$K_m = \frac{k U_* G}{f} \quad (28)$$

## 2. Cardone's Extension of Blackadar's Two-Layer Model

The success of Blackadar's two-layer model in predicting the essential characteristics of the neutral surface boundary layer, along with its relative ease of application, suggested its extension to the non-neutral marine boundary layer. Cardone extended the diabatic model to over-water flow by regarding  $z_o$  as a function of  $U_*$ .

The eddy viscosity in the constant stress layer for the non-neutral case was expressed by Cardone in terms of the nondimensional wind shear as

$$K_m = \frac{k U_* z}{\theta_u (z/L')} \quad (29)$$

Since  $K_m$  is constant with height above  $h$ , its value in the Ekman layer is then

$$K_m = \frac{k U_* B_o G}{f \beta_u (z/L')} . \quad (30)$$

The wind speed at  $h$  was given by

$$U_h = \frac{U_*}{k} \left[ \ln B_o R_o - \Psi(h/L') \right] . \quad (31)$$

Internal boundary conditions applied at level  $h$  appear in Figure 3 and follow from the fact that  $\frac{\partial W}{\partial z}$  always makes an angle of  $\frac{3\pi}{4}$  to the ageostrophic wind ( $W$ ) and that  $\frac{\partial U}{\partial z}$  is parallel to  $U$  in the constant stress layer.

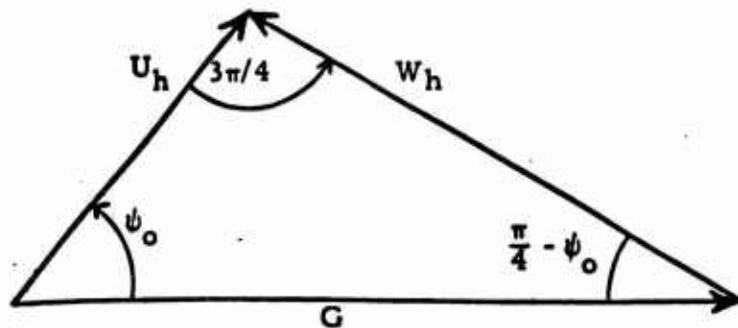


Fig. 3. Wind vector diagram applicable at level  $h$ .

By requiring continuity of wind shear at  $h$ , as well as directional continuity, Cardone was able to reduce the problem (analogous to (26) and (27)) to

$$\frac{U_*}{G} = \left[ 2 k B_o \sin^2 \Psi_o \beta_u (h/L') \right]^{1/2} \quad (32)$$

$$\frac{U_*}{G} = \frac{k \sqrt{2} \sin (\pi/4 - \Psi_o)}{\ln B_o R_o - \Psi(h/L')} . \quad (33)$$

These equations are solved simultaneously for  $\frac{U_*}{G}$  and  $\Psi_o$  from the external parameters  $G$ ,  $f$ , and  $(\theta_a - \theta_s)$ . In addition, a dimensionless stability parameter was defined as

$$L_* = \frac{B_o G}{f L'} \quad (34)$$

which represented the ratio of the surface layer  $h$  to the stability length  $L'$  (Blackadar and Ching, 1965).

The relationships determined by Cardone for  $U_*/G$  and  $\Psi_o$  as functions of  $R_o$  and  $L_*$  appear in Figures 4 and 5. For all surface Rossby numbers,  $U_*/G$  increases and  $\Psi_o$  decreases with decreasing stability. In general, stable conditions have a more pronounced effect on the surface layer wind characteristics than unstable conditions. In comparing the results from the model with observations, Cardone noted that  $U_*/G$  increased by roughly 30% in unstable conditions, with no marked tendency to decrease at great instability, and decreased roughly 70% in stable conditions. These features in the observed results will be discussed later.

The effects of baroclinicity were also included by assuming that the geostrophic wind was a linear function of height (Blackadar, 1965b). Blackadar (1965a) introduced a parameter,  $p$ , which was defined at level  $h$  by

$$p = \frac{U_h}{(\partial U / \partial z)_h} = \frac{B_o G}{f \beta_u (h/L')} \left[ \ln B_o R_o - \Psi(h/L') \right]. \quad (35)$$

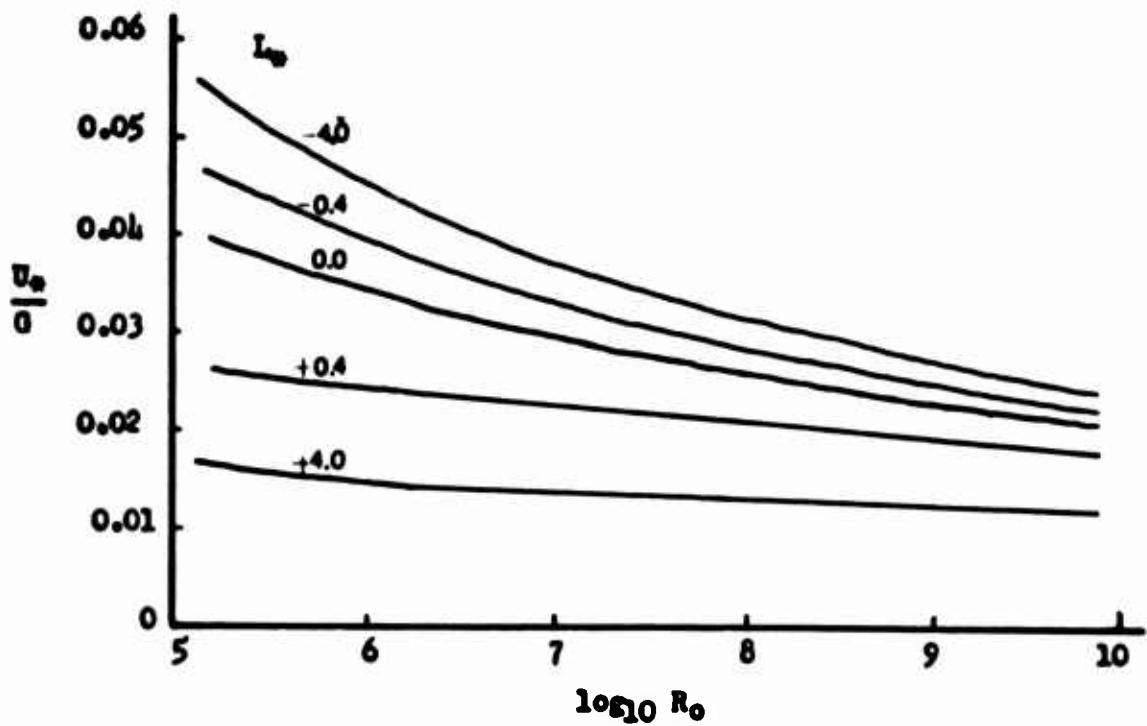


Fig. 4. Variation of the geostrophic drag coefficient with surface Rossby number and dimensionless stability length ( $L_s$ ) (Cardone, 1969)

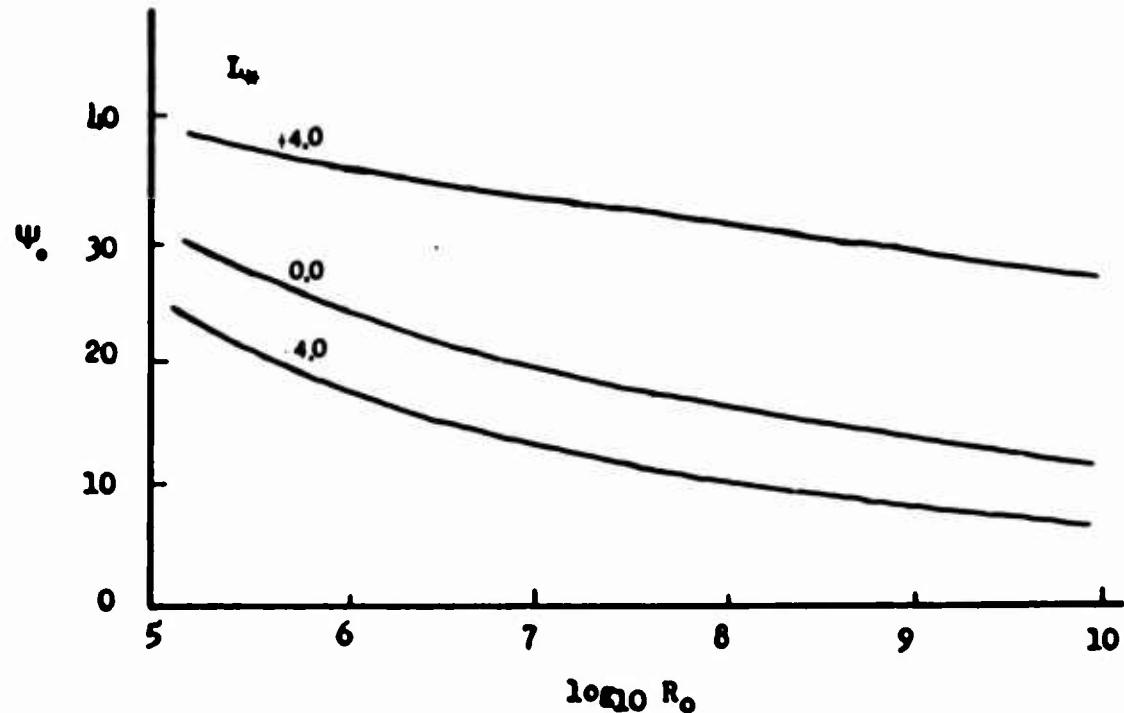


Fig. 5. Variation of surface cross isobar angle with surface Rossby number and stability (Cardone, 1969)

The requirement of continuity of wind shear for the baroclinic case took the form

$$U_h = \bar{p} \left[ \left( \frac{\partial G}{\partial z} \right)_h + \frac{\partial W}{\partial z} \right]_h . \quad (36)$$

Figure 6 illustrates the geometrical implication of (36). From the Ekman layer solution of the ageostrophic wind, the following equation can be obtained

$$\bar{p} \left| \frac{\partial W}{\partial z} \right|_h = \sqrt{2} p' |W_h| , \quad (37)$$

where  $p'$  is a dimensionless parameter defined by

$$p' = \frac{B_o G}{g_u (h/L')} \sqrt{\frac{g_u (h/L')}{2 k U_* B_o G}} \left[ \ln B_o R_o - \Psi (h/L') \right] . \quad (38)$$

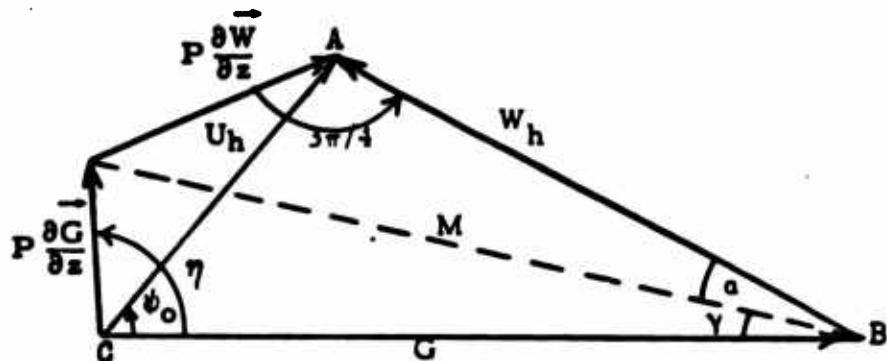


Fig. 6. Geometrical relationships resulting from the requirement of the continuity of wind shear at the internal boundary level  $h$  (Blackadar, 1965b).

Next the thermal wind vector was replaced by its dimensionless magnitude

$$r = \frac{1}{f} \left| \frac{\partial G}{\partial z} \right| \quad (39)$$

where the direction of  $\frac{\partial G}{\partial z}$  relative to the surface geostrophic wind was implied by angle  $\eta$  in Figure 6. Still another dimensionless quantity was defined

$$r' = r B_o [ \ln B_o R_o - \Psi(h/L') ] \quad (40)$$

and (35) was then rewritten as

$$p \left| \frac{\partial G}{\partial z} \right| = r' G. \quad (41)$$

By applying the law of cosines to each of the two smaller triangles in Figure 6 that have  $M$  for one side, the following relations were determined

$$M^2 = |w_h|^2 s^2 = G^2 q^2 \quad (42)$$

$$s^2 = (1 + 2p' + 2p'^2) \quad (43)$$

$$q^2 = (1 + r'^2 - 2r' \cos \eta). \quad (44)$$

$\eta$  is the angle between the geostrophic wind and the thermal wind (see Figure 6). Then the law of sines was applied to yield

$$\sin \alpha = p'/s \quad (45)$$

$$\sin \gamma = \frac{r'}{q} \sin \eta \quad (46)$$

The following relations apply to the large triangle (ABC) in Figure 6.

$$U_h = |w_h| \sin (\alpha + \gamma) / \sin \Psi_o \quad (47)$$

$$U_h^2 = G^2 + |w_h|^2 - 2G |w_h| \cos (\alpha + \gamma) \quad (48)$$

$$|w_h|^2 = U_h^2 + G^2 - 2 U_h G \cos \Psi_o. \quad (49)$$

From equations (31), (45), (46), (47), (48), and (49), the following equations describing the surface boundary layer characteristics were obtained

$$\tan \Psi_0 = \frac{\sin(\alpha + \gamma)}{(s/q) - \cos(\alpha + \gamma)} \quad (50)$$

$$U_* = \frac{k G q \sin(\alpha + \gamma)}{s \sin \Psi_0 [\ln B_0 R_0 - \Psi(h/L')]} \quad (51)$$

By including (23) and (15), Cardone obtained a system of equations that could be solved for  $U_*$  and  $\Psi_0$  from the input parameters  $G$ ,  $f$ ,  $(\theta_a - \theta_s)$ ,  $\left| \frac{\partial G}{\partial z} \right|$ , and  $\eta$ . These equations were calculated from a knowledge of sea level pressure, surface air temperature, sea surface temperature, and latitude.

### 3. Some Characteristics of the Marine Surface Boundary Layer

A primary effort of Cardone's work was to determine how well the planetary boundary layer model, using only large-scale synoptic parameters as input, specified the surface boundary layer wind distribution over the sea surface. Cardone considered several studies relating the geostrophic wind ratio  $U_z/G$  and the air-sea temperature difference. The results of an extensive study by Carstensen (1967) at Fleet Numerical Weather Central (FNWC) appear in Figure 7 along with earlier results due to Bleeker. The ratio of surface wind ( $V$ ) to geostrophic wind ( $G$ ) is tabulated for each degree of the reported air-sea temperature difference, and the median value for each interval is plotted as an  $x$  in Figure 7. The validity of the results was limited to the air-sea temperature difference range  $-4$  to  $+1^\circ\text{C}$ . In order to include observed winds in the surface pressure analysis, FNWC uses a constant value for  $V/G$  of 0.78 and a turning angle of  $15^\circ$  toward lower pressure. This selection could produce significant error.

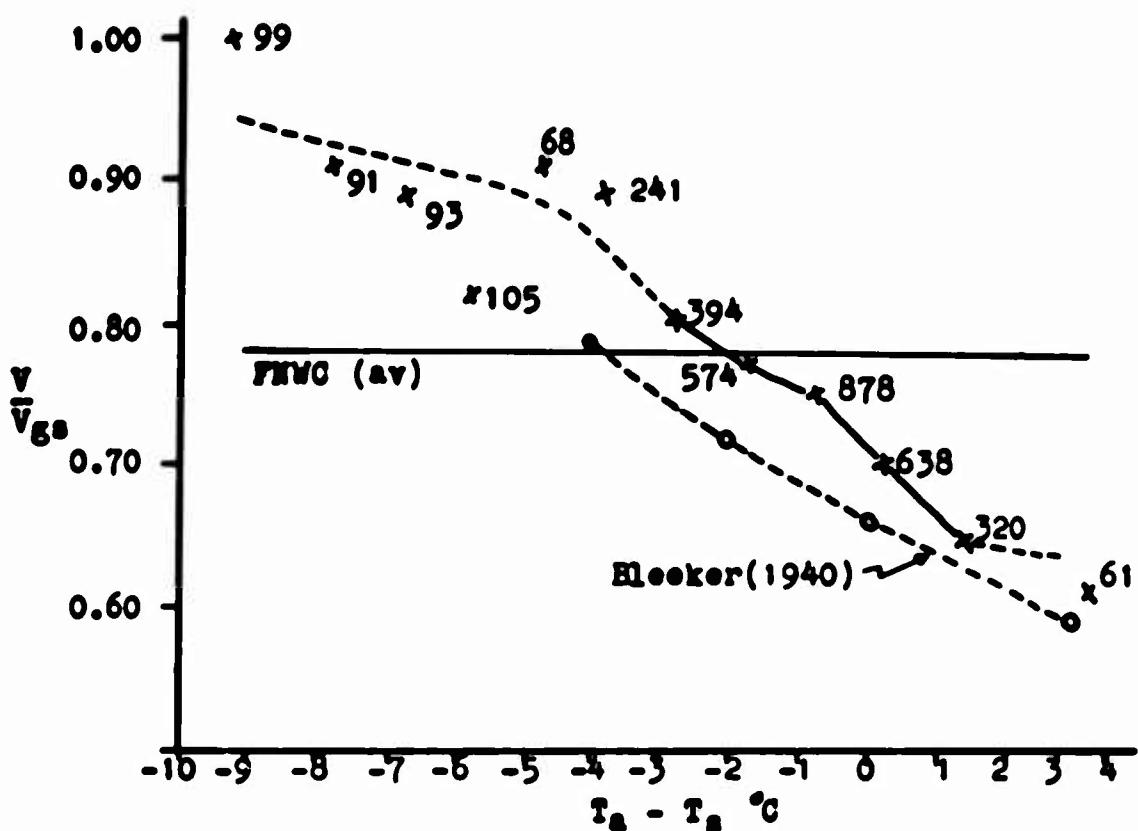


Fig. 7. Reported wind-geostrophic wind ratio as a function of air-sea temperature difference (Carstensen, 1967)

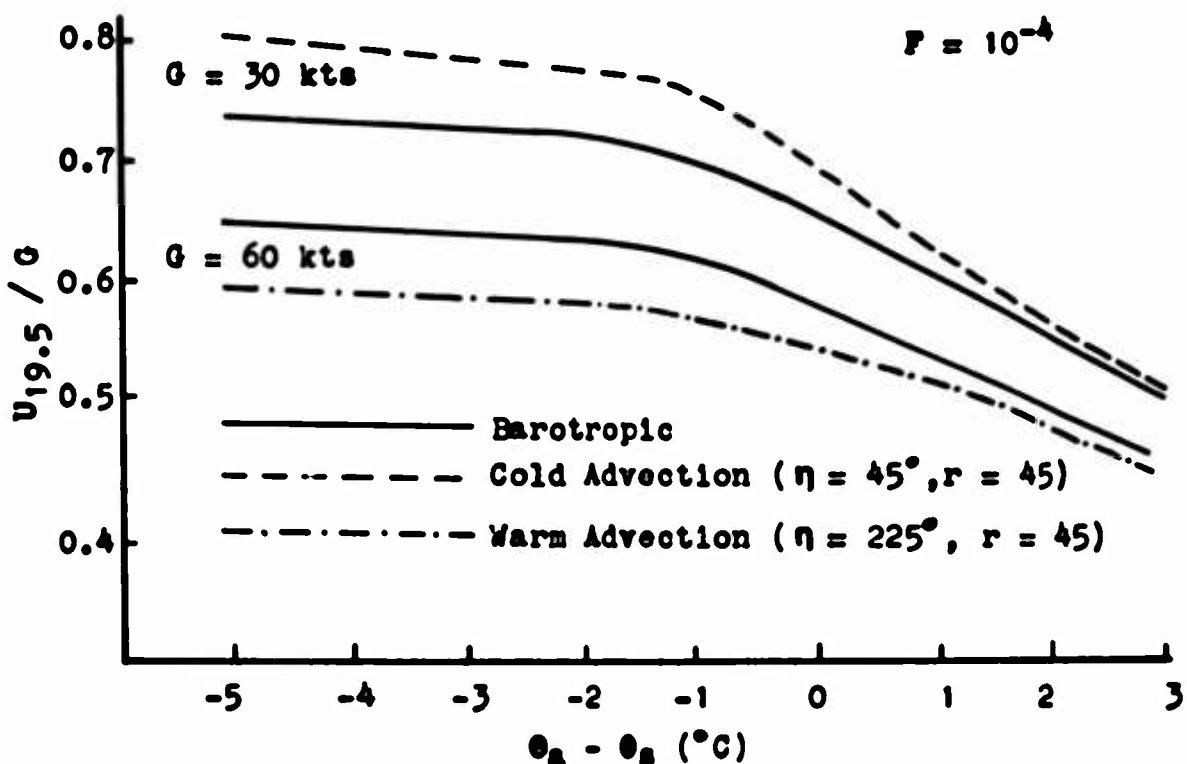


Fig. 8. Two layer model predictions of the dependence of the ratio of the 19.5 meter wind speed to the geostrophic wind speed upon air-sea temperature difference and thermal advection

Cardone's model can be used to compute a wind speed at 19.5 meters. Figure 8 illustrates the variation of the ratio of the 19.5 meter wind and surface geostrophic wind with the air-sea temperature difference. The ratio varied less in unstable conditions than in stable conditions.

The effects of baroclinicity also appear in Figure 8 and these effects increase as stability decreases, with the ratio,  $U_{19.5}/G$ , increasing with cold advection and decreasing with warm advection. Mendenhall (1967) also studied the influence of baroclinicity and stability on the veering of the wind in the planetary layer. He concluded that over mid-latitude oceans the diurnal variation of frictional veering of the wind with height was negligible, but that lapse rate and baroclinicity were important. Results obtained by Cardone, using the planetary model, showed qualitative agreement with Mendenhall's conclusions. Cardone's results also suggested that baroclinicity was more significant than stability in determining the veering of the wind in the planetary boundary layer. For both cases in situations of strong, cold advection the actual wind may back with height, especially in conjunction with unstable stratification.

Additional results on the influence of baroclinicity came from a study by Clarke (1970), who found a tendency for the boundary layer wind to back instead of veer with height in unstable conditions. He stated that this could have been due to baroclinicity or to acceleration. He noted that it could have occurred if  $\frac{\partial T^*}{\partial z}$  was systematically

negative in convective conditions up to a level of about  $0.2 \frac{U_*}{f}$  instead of being positive as in the normal case. This latter interpretation was based on the observation that the wind appeared to be sub-geostrophic in the convective layer and reached geostrophic values only after a rapid increase through the overlying inversion.

### III. COMPUTATIONS AND NUMERICAL PROCEDURES

#### A. EXTERNAL WIND FIELD SPECIFICATIONS

All computations were made with data defined on a subset of the FNWC 63X63 grid which covered the North Pacific Ocean (Figure 9). Data fields included sea level pressure, sea surface temperature, and surface air temperature at each of 315 grid points spaced 381 km apart at 60°N. The necessary map factor was included for a 1:30 million polar stereographic projection.

The coriolis term appears as an external parameter and therefore poses a limitation on the lowest geographical latitude where the geostrophic relation would be valid. It has already been established that the dynamic boundary layer was formed by the surface stress and the coriolis acceleration. Pushistov (1970) concluded that manifestations of the nonlinearity in the dynamics of the boundary layer and the appearance of a perceptible ageostrophic component of velocity in the free atmosphere were to be expected near latitudes of 7-10°. Sheppard (1970) suggested 20°N as a limiting latitude for application to boundary layer models. A limiting value of 15° latitude was used in this program as the limit on geostrophic flow. For latitudes below 15°, the coriolis parameter would be kept constant.

##### 1. Geostrophic Wind Computations

The geostrophic wind components were computed from the surface pressure field (P) as follows

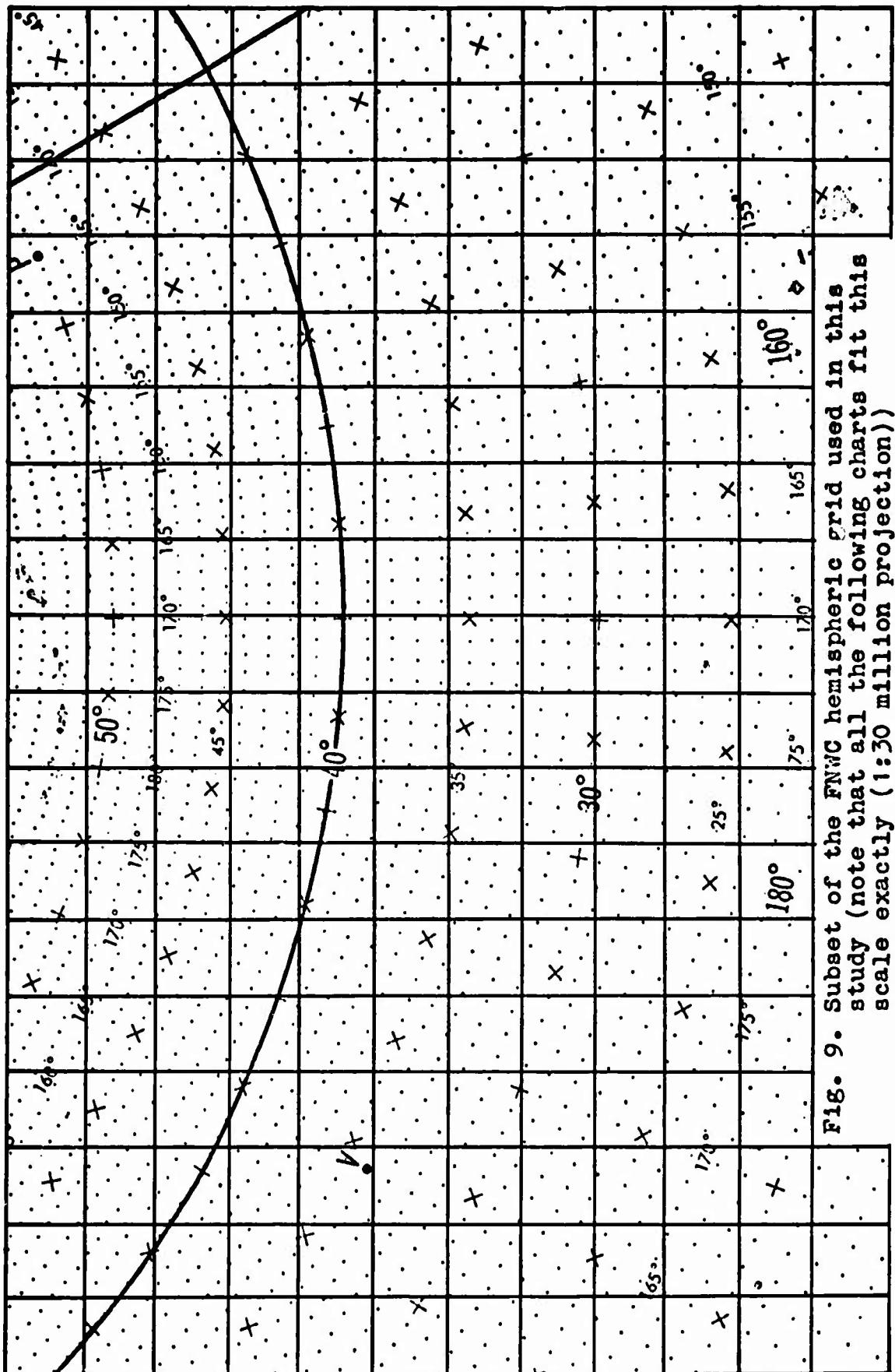


FIG. 9. Subset of the FNMNC hemispheric grid used in this study (note that all the following charts fit this scale exactly (1:30 million projection))

$$u_g = - \frac{1}{\bar{p}_a f} \frac{\partial P}{\partial y} \quad (52)$$

$$v_g = \frac{1}{\bar{p}_a f} \frac{\partial P}{\partial x} \quad (53)$$

using a centered explicit finite difference technique. The density was determined from the equation of state

$$\bar{p}_a = \frac{P}{R_d \theta_a} \quad (54)$$

where  $R_d$  is the gas constant for dry air, and  $\theta_a$  the surface air temperature in degrees Kelvin. The magnitude of the geostrophic wind was computed at each grid point from

$$|G| = \sqrt{u_g^2 + v_g^2} \quad . \quad (55)$$

## 2. Correction for Curvature

A gradient wind was calculated to correct for the effect of curvature on the surface geostrophic wind. Two basic principles were required for this correction. The first was that following an air parcel along an isobar, the total derivative was zero or

$$dP(x, y) = 0. \quad (56)$$

The second requirement was the mathematical statement of curvature

$$K_c(x, y) = \frac{y''}{[1 + (y')^2]^{3/2}} = \frac{1}{R} \quad (57)$$

where  $y'$  and  $y''$  were the first and second derivatives with respect to  $x$ , and  $R$  was the radius of curvature. By expanding the total derivative of

P, obtaining  $y'$  and  $y''$ , and rearranging, the following result was obtained for the radius of curvature

$$R = \frac{\left(\frac{\partial P}{\partial y}\right)^2 \left[1 + \left(\frac{\partial P/\partial x}{\partial P/\partial y}\right)^2\right]^{3/2}}{\frac{\partial P}{\partial x} \left[2 \frac{\partial^2 P}{\partial x \partial y} - \frac{\partial^2 P}{\partial y^2} \frac{\partial P/\partial x}{\partial P/\partial y}\right] - \frac{\partial P}{\partial y} \frac{\partial^2 P}{\partial x^2}}. \quad (58)$$

Next the gradient wind was computed from

$$V_G = G - \frac{V^2}{fR}. \quad (59)$$

Solving for  $V_G$  and rearranging led to a workable form of the equation

$$V_G = \frac{G}{1/2 + \sqrt{1/4 + G/Rf}}. \quad (60)$$

For this case the magnitude of the term  $G/Rf$  determined how much curvature would be allowed. The maximum values were chosen to be  $-3/16$  for anticyclonic curvature and  $+3/4$  for cyclonic curvature which yielded a range of  $V_G = 1.33G$  for anticyclonic curvature and  $V_G = 0.67G$  for cyclonic curvature. This agreed with generally accepted synoptic limits.

Subroutine GRADWD (Appendix A) makes the necessary computations and follows a similar scheme utilized by FNWC (Kaitala, 1971 unpublished). Centered finite difference schemes were used to calculate the derivatives.

### 3. Thermal Wind Correction

The nondimensional thermal wind components were determined from

$$u_T = - \frac{g}{f^2 \frac{\theta}{\theta_a}} \frac{\partial \theta}{\partial y} \quad (61)$$

$$v_T = \frac{g}{f^2 \theta_a} \frac{\partial \theta}{\partial x} , \quad (62)$$

and the total thermal wind was given by

$$r = \sqrt{u_T^2 + v_T^2} . \quad (63)$$

For this model,  $r$  had a limiting value of 100. The geostrophic and thermal wind angles were calculated from the previously computed wind components. This procedure was repeated for each grid point and was used as input data to the boundary layer model.

## B. PROCEDURES IN SOLVING THE BOUNDARY LAYER EQUATIONS

### 1. Initial Conditions

#### a. Geostrophic Wind Speed Less Than Or Equal to 5.0 meters/second:

This case was associated with the aerodynamically smooth flow region described earlier. For this case no corrections were made for stability and a constant drag coefficient of 0.022 was assumed.

This resulted in a constant inflow angle of  $15^\circ$ . The 19.5 meter wind was determined from

$$U_{19.5} = 0.7 G \quad (64)$$

The choice of the drag coefficient was not arbitrary, but depended on the relationship between  $C_z$  and  $U_z$  (see Figure 2). The value of  $C_{10}$  was a minimum at 6 meters/second under neutral conditions. The value of  $C_{19.5}$  was extrapolated from this curve.

b. Geostrophic Wind Speed Greater Than 5.0 meters/  
second and Absolute Value of Air-Sea Temperature  
Difference Less Than or Equal to 1°C:

Initially the drag coefficient was  $C_z = 0.0245$ , with the thermal wind ( $r$ ) and the angle between the geostrophic and thermal wind ( $\eta$ ) set equal to zero. For this case  $L_*$ ,  $L'$  and  $\Psi(h/L')$  were set equal to zero and  $\phi_u(h/L') = 1.0$ . A solution was then obtained using the modified baroclinic model of Blackadar described below.

c. Geostrophic Wind Speed Greater Than 5.0 meters/  
second and Absolute Value of Air-Sea Temperature  
Difference Greater than 1°C:

The drag coefficient  $C_z = 0.0245$  is used to estimate the initial  $U_*$  and  $r$  and  $\eta$  are set equal to zero.  $\theta_a - \theta_s$  was restricted to the range -15 to +4.0°C. The values of  $L_*$ ,  $L'$ ,  $\Psi(h/L')$ , and  $\phi_u(h/L')$  were computed from  $z_a$ ,  $\theta_a - \theta_s$  and  $U_*$  and a solution was obtained using the modified Blackadar model described below.

$L_*$  was computed from (34) and  $L'$  from (15) using as a first approximation

$$\Psi(z_a/L') = 10 - \ln(z_a/z_0). \quad (65)$$

$\Psi(h/L')$  was computed from SUBROUTINE PSI (Appendix A). If  $L_*$  was greater than zero (stable case), then

$$\Psi(h/L') = 0.7 L_* \quad (66)$$

If  $L_*$  was less than or equal to zero (unstable case), a value of Richardson number and  $\theta_u(h/L')$  had to be computed. This was accomplished in SUBROUTINE SHR (Appendix A). Then  $\Psi(h/L')$  was calculated from (13).

$\theta_u(h/L')$  was computed using SUBROUTINE SHR (Panofsky, 1968; Cardone, 1969). If  $L_*$  was greater than zero (stable case), then

$$\theta_u(h/L') = 1 + 7 L_* \quad (67)$$

If  $L_*$  was less than or equal to zero (unstable case), then a new Richardson number was determined from

$$R_{I_{\text{new}}} = L_* (1 - 18 R_{I_{\text{old}}})^{1/4} \quad (68)$$

where initially  $R_{I_{\text{old}}} = L_*$ . This iteration was continued until the inequality

$$\left| R_{I_{\text{new}}} - R_{I_{\text{old}}} \right| < \epsilon \quad (69)$$

was satisfied, where  $\epsilon = 0.01$  for this model. Then  $\theta_u(h/L')$  was found from

$$\theta_u(h/L') = \frac{1}{(1 - 18 R_{I_{\text{new}}})^{1/4}} \quad (70)$$

## 2. The Modified Blackadar Baroclinic Boundary Layer Model

The basis of this model was an iteration procedure which included the effects of stability and baroclinicity on the computed wind. The procedure was free of computational instability and converged rapidly for all reasonable choices of initial  $U_*$ . First values of  $z_0$ ,

$p$ ,  $p'$ ,  $s$ ,  $q$ ,  $\alpha$ ,  $\gamma$ ,  $\Psi_o$ , and  $U_{*_{new}}$  were obtained using respectively equations (23), (35), (38), (43), (44), (45), (46), (50), and (51).

The iterative procedure was repeated until the inequality

$$\left| U_{*_{new}} - U_{*} \right| \leq \epsilon \quad (71)$$

was satisfied where  $\epsilon$  was 0.05. The iteration was repeated, setting  $U_{*} = U_{*_{new}}$ , until the inequality was again satisfied. If  $L_{*}$  was not zero, then new values of  $\Psi$ ,  $z_o$ , and  $L'_{new}$  were computed where the argument of  $\Psi$  was  $z_a/L'$ . If

$$\left| L'_{new} - L' \right| \geq 5 \quad (72)$$

then  $L'$  was set equal to  $L'_{new}$ . The model was repeated from where  $L_{*}$  was first calculated until the inequality was satisfied. After completing the iterations for the case where the thermal wind ( $r$ ) and  $\eta$  were equal to zero, the computed values of  $r$  and  $\eta$  were substituted and the model was again repeated until all criteria were satisfied. If  $L' \neq 0$ , then  $\Psi(z/L')$  was computed as  $\Psi(19.5/L')$ ; otherwise  $\Psi(19.5/L')$  was set equal to zero. Finally the wind at 19.5 meters was determined from

$$U_{19.5} = \frac{U_{*}}{k} \left[ \ln \frac{19.5}{z_o} - \Psi \left( \frac{19.5}{L'} \right) \right] \quad (73)$$

Values of  $\theta_a - \theta_s$ ,  $G$ ,  $z_o$ ,  $z/L'$ ,  $\Psi_o$ ,  $U_{*}$ ,  $U_{19.5}$ ,  $\log R_o$ ,  $U_{*}/G$ ,  $U_{19.5}/G$  and  $R_i$  were then available for each grid point.

### C. COMPUTATION OF INDEX OF REFRACTION CONSTANT

In this section possible relationships are considered between the refractive index structure constant and quantities dependent only on the wind speed and air-sea temperature difference. Specifically those quantities that can be predicted from the model considered in this study are examined.

The propagation of light waves and radio waves is affected primarily by air-density inhomogeneities and therefore by temperature fluctuations. One effect of these fluctuations is scintillation or rapid changes in intensity at a point. These are distortions caused by variations in the speed of propagation through turbulent elements having different densities, and hence different indexes of refraction. The refractive index structure constant ( $C_n^2$ ) is the primary parameter characterizing the turbulent fluctuations of the atmospheric refractive index for scales from a few millimeters to several meters. If the quantity  $C_n^2$  is known, calculations can be made on scintillation, beam spreading, and beam wander, all of which are important in describing light wave propagation.

A relation necessary in the formulation is the one-dimensional spectrum for temperature fluctuations in the inertial subrange described by Corrsin (1951)

$$S_0 (K) = B \epsilon^{-1/3} N K^{-5/3} \quad (75)$$

where  $\epsilon$  is the rate of dissipation of turbulent kinetic energy,  $N$  is a measure of the "smearing" of temperature inhomogeneities,  $B$  is an

empirical constant (= 0.81, Boston, 1970), and  $K$  is the streamwise component of wave number. The structure constant for temperature fluctuations ( $C_T^2$ ) is defined as

$$C_T^2 = B \epsilon^{-1/3} N \quad (76)$$

and is related to the index of refraction as follows

$$C_n^2 = \left( \frac{-79 \times 10^{-6} \rho}{T_a^2} \right)^2 C_T^2. \quad (77)$$

Another expression which has to be considered is the nondimensional balance equation for turbulent kinetic energy, which, for stationary, horizontally homogeneous flow is (Busch and Panofsky, 1968)

$$\beta_u - z/L - \beta_c - \beta_D = 0. \quad (78)$$

$\beta_u$  is the nondimensional wind shear (8).  $\beta_D$  represents the divergence terms, and

$$\beta_c = - \frac{\epsilon k z}{U_*^3}. \quad (79)$$

The following relations arise from the definitions in the nondimensional turbulent kinetic energy balance and definition of  $N$

$$\epsilon = \beta_c \frac{U_*^3}{k z} \quad (80)$$

$$N = -\bar{\theta}w \frac{\partial \theta}{\partial z} = K_H \left( \frac{\partial \theta}{\partial z} \right)^2. \quad (81)$$

Substitution of these two expressions, plus the definition of  $K_m$  (29)

into the definition of  $C_T^2$  yields (Panofsky, 1968)

$$C_T^2 = B \beta_c^{-1/3} \left( \frac{K_H}{K_m} \right) \frac{1}{\beta_u} (k z)^{4/3} \left( \frac{\partial \theta}{\partial z} \right)^2. \quad (82)$$

Equation (82) is not a suitable expression because of the lack of information on  $\beta_c$ . However if the effect of the divergence terms is neglected,  $\beta_c$  can be expressed in terms of  $\beta_u$  and  $z/L$  as

$$\beta_c = \beta_u - z/L. \quad (83)$$

It is noted that with wind-wave interaction, the assumption that the divergence terms can be neglected is probably not valid, but the possible effect this assumption has on predictions of  $C_T^2$  has not been studied.

Substitution of (83) into (82) yields

$$C_T^2 = B \left[ \beta_u - z/L \right]^{-1/3} \left[ \frac{K_H}{K_m} \frac{1}{\beta_u} \right] (kz)^{4/3} \left( \frac{\partial \theta}{\partial z} \right)^2. \quad (84)$$

This final equation for  $C_T^2$  lends itself to the boundary layer model because all the parameters can be computed within the model. Thus, the model can be used to obtain a horizontal distribution of  $C_T^2$  over water for various wind and stability conditions. The average stability  $\partial \theta / \partial z$  is determined from (14).

The ratio of the exchange coefficients  $K_H / K_m$  is set equal to 1.0 in Cardone's formulation. However it has been shown that  $K_H$  and  $K_m$  are not usually identical except for small values of  $\partial \theta / \partial z$  for which the turbulent motion is not significantly influenced by buoyancy forces. Businger et al. (1971) suggested the following form for the ratio of eddy diffusions

$$a_h = 1.35 \frac{(1 - 9z/L)^{1/2}}{(1 - 15z/L)^{1/4}}. \quad (85)$$

Here  $a_h = 1.35$  for neutral conditions, which agrees closely with laboratory measurements (cf., Hinze, 1959). A value of 1.35 rather than 1.0 for  $a_h$  under neutral conditions arises because the von Kármán constant was equal to 0.35 instead of the usual 0.40 (Businger et al., 1971).

#### D. WAVE HEIGHT ANALYSIS

Wave analysis is another application for the boundary layer model. For this purpose, a simplified version of the FNWC sea-swell model (Hubert and Mendenhall, 1970) was utilized. The u and v components of the 19.5 meter wind determined from the boundary layer model were used as both current and history input in this analysis.

The FNWC sea-swell model computes a weighted wind from the wind components as follows:

$$u = 0.3 u_{t-12} + 0.7 u_t$$

$$v = 0.3 v_{t-12} + 0.7 v_t \quad (86)$$

where  $t-12$  is the 12-hour old history wind speed and  $t$  is the current wind speed. These component values are saved as history for the next analysis run, and then the total wind ( $V$ ) is computed. The wave heights ( $W_A$ ) in feet are then computed from

$$W_A = 7 \left[ \left( \frac{v}{10} \right)^2 - \left( \frac{v}{20} \right)^{3.3} \right] + 1.0 \quad (87)$$

Although this is a very simple empirical method for determining wave heights, the basic method has been used by FNWC since the mid-1960's and this particular model has been used operationally since August 1970.

In addition, the FNWC model computed corrections for fetch and ice, and performed some smoothing of the resulting height field. It was assumed these corrections would be minimal for this application.

#### IV. SPECIFICATION OF EXTERNAL PARAMETERS

A primary objective of this study was to apply the two-level baroclinic boundary layer model to actual atmospheric conditions and compare the computed results with the observed results. The time period 00 GMT 28 November 69 through 00 GMT 02 December 69, with an analysis interval of every 12 hours, was chosen for this comparison for several reasons. First an intense low pressure system, which influenced most of the central North Pacific Ocean, developed during this period with the system reaching maximum intensity about 00 GMT 30 November 69. The system provided input data which allowed the model to predict boundary layer parameters under varying atmospheric conditions. Secondly considerable synoptic data had been gathered by the author during an earlier study on the extreme sea conditions (unpublished) associated with this storm. The latter gave a good basis for comparing computed values to observed data. Finally, this storm produced waves and swell which caused considerable damage to several islands of the Pacific as well as to the spar buoy FLIP. Therefore a wave generation model was applied using the predicted surface winds, and these predicted waves were compared to the observed wave conditions.

The 00 GMT 30 November 69 period was of the most interest due to its extreme conditions, extensive data coverage, and it was the probable period when the destructive waves were being generated. The various corrections described above were incorporated into the model for this

time period, and synoptic data were plotted and analyzed for surface pressure, air-sea temperature difference, surface wind speed and wave height.

Surface layer parameters were computed including various combinations of stability, curvature, and thermal wind to yield information on the relative importance of each parameter in the formulation. The following notation will be used in describing the results:

NN - non-neutral, correction for stability

N - neutral - no stability correction

NC - no curvature correction applied

C - curvature correction applied

NT - no thermal wind correction applied

T - thermal wind correction applied

It was recognized at the onset of the study that the specification of the external fields ( $\Delta\theta$  and  $G$ ) had to be reasonably accurate in order to evaluate the model on the basis of observed wind fields. For example, if the observed surface winds were quite different than those predicted by the model, some estimate had to be available on the possibility that the discrepancies were due to poor specification of the geostrophic wind or the air-sea temperature difference. Therefore in this section consideration is given to the relevant aspects of representing those fields which make up the external parameters.

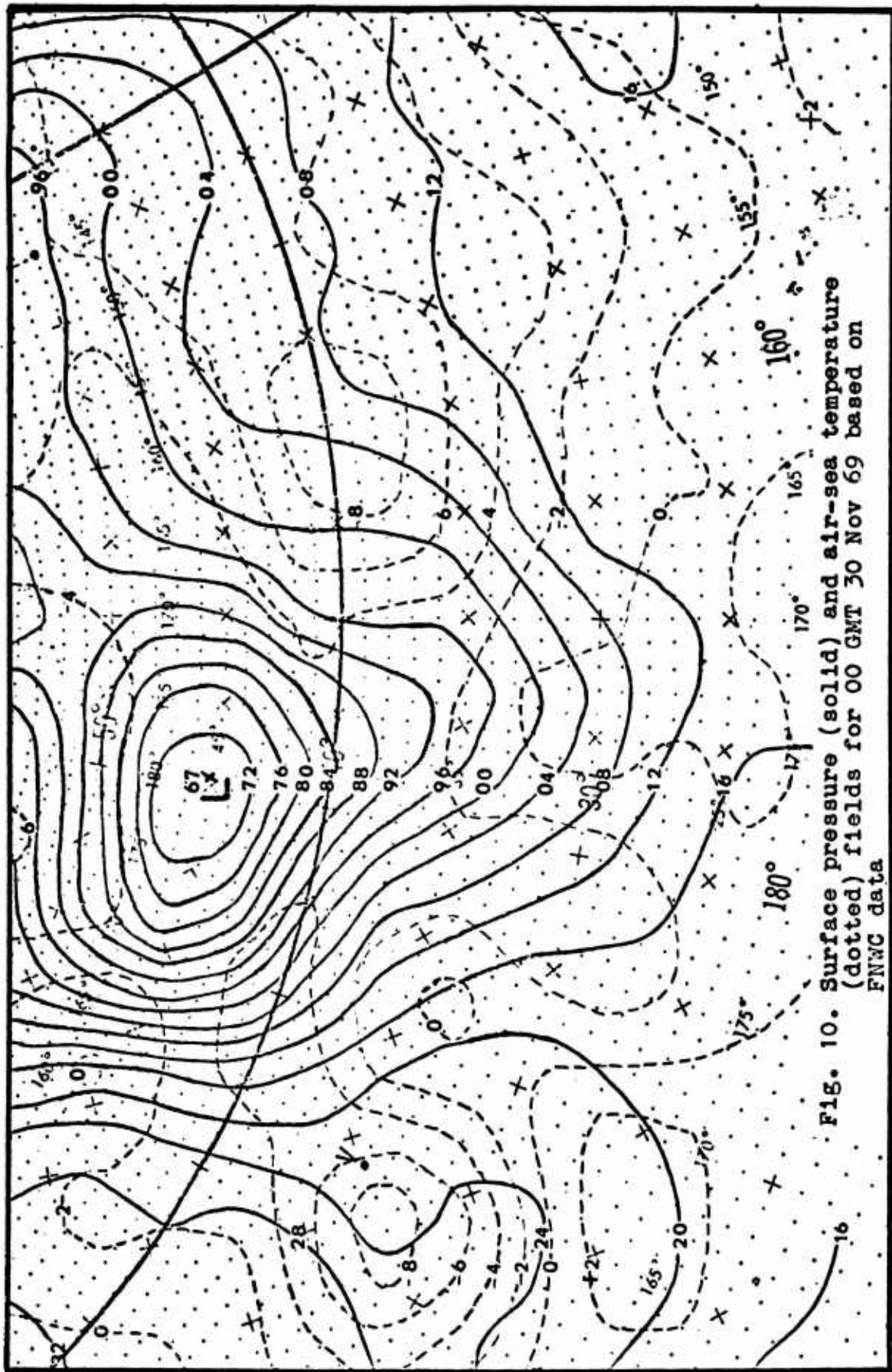
The charts in the following figures correspond to a 1:30 million polar stereographic projection with the grid as depicted in Figure 9.

## A. FNWC ANALYSIS

Sea level pressure and air-sea temperature difference fields for 00 GMT 30 November appear in Figure 10. A 967 mb low pressure center was located near 45°N 180° with considerable thermal instability both ahead of and behind the system. A significant feature of this system was the intense isobaric gradient in the western quadrant.

The first consideration was the comparison of the computed surface geostrophic wind with the computed gradient wind for this period. The geostrophic and gradient wind fields computed from the surface pressure fields in Figure 10 appear in Figures 11 and 12 (units of meters/second). Figure 13 depicts the influence of the curvature correction on the geostrophic wind computed as a percent difference from geostrophic wind minus gradient wind then divided by geostrophic wind.

The two maximum wind centers west and north of the Low are based on data at one grid point and result from the influence of anticyclonic curvature, strong isobaric gradients, and finite difference computations. These features do not appear to be realistic. However, the wind speed distribution depicted by them is good except that the magnitude of the wind speed may be a little too large. It was expected that the maximum change would occur in an area of maximum curvature and maximum pressure gradient from the relation in (60). Generally speaking, the change in wind speed due to the curvature correction is 10 to 20% with a maximum change of 34% imposed by the limits placed on the term  $G/Rf$  in (60).



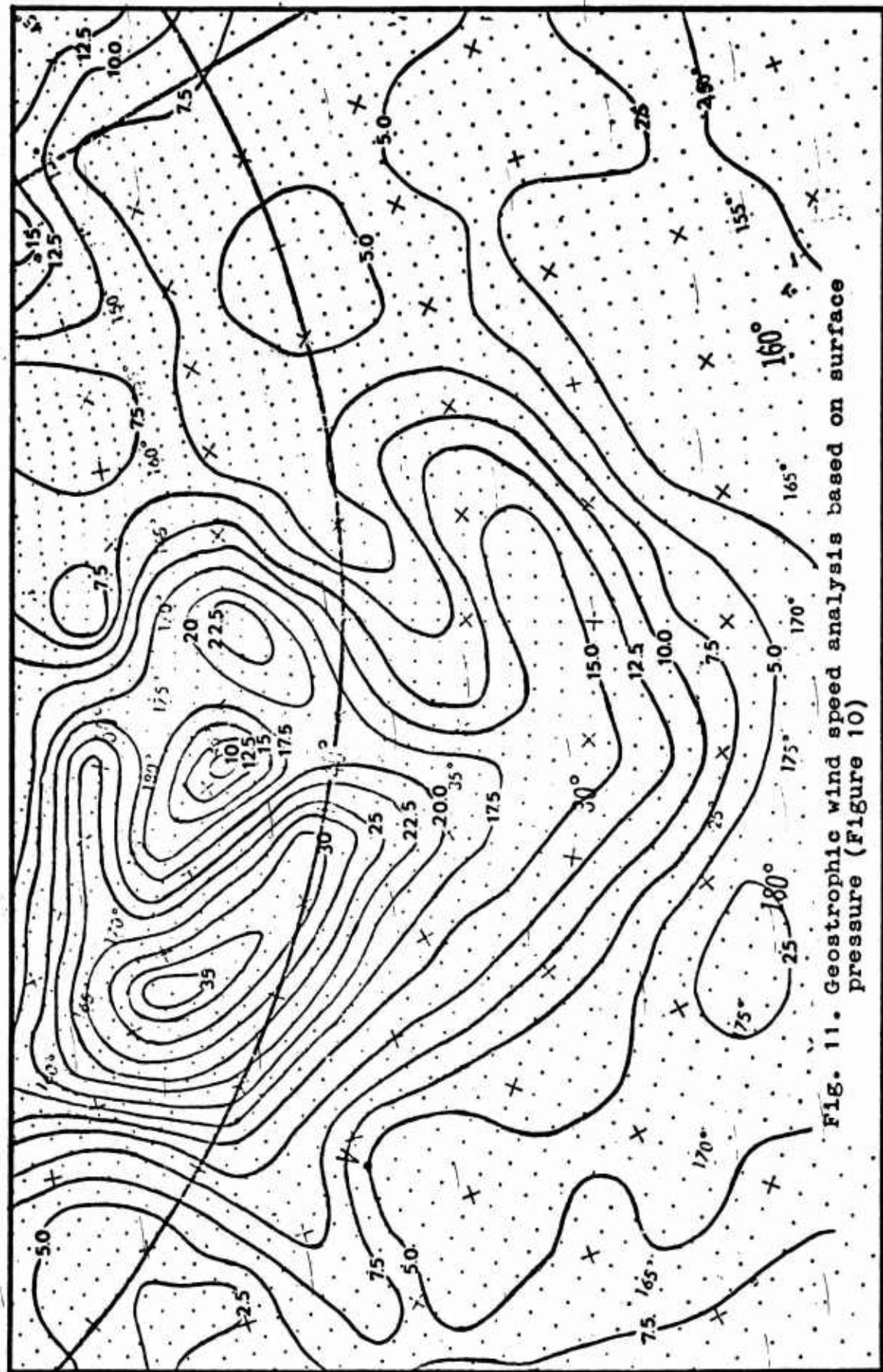


Fig. 11. Geostrophic wind speed analysis based on surface pressure (Figure 10)

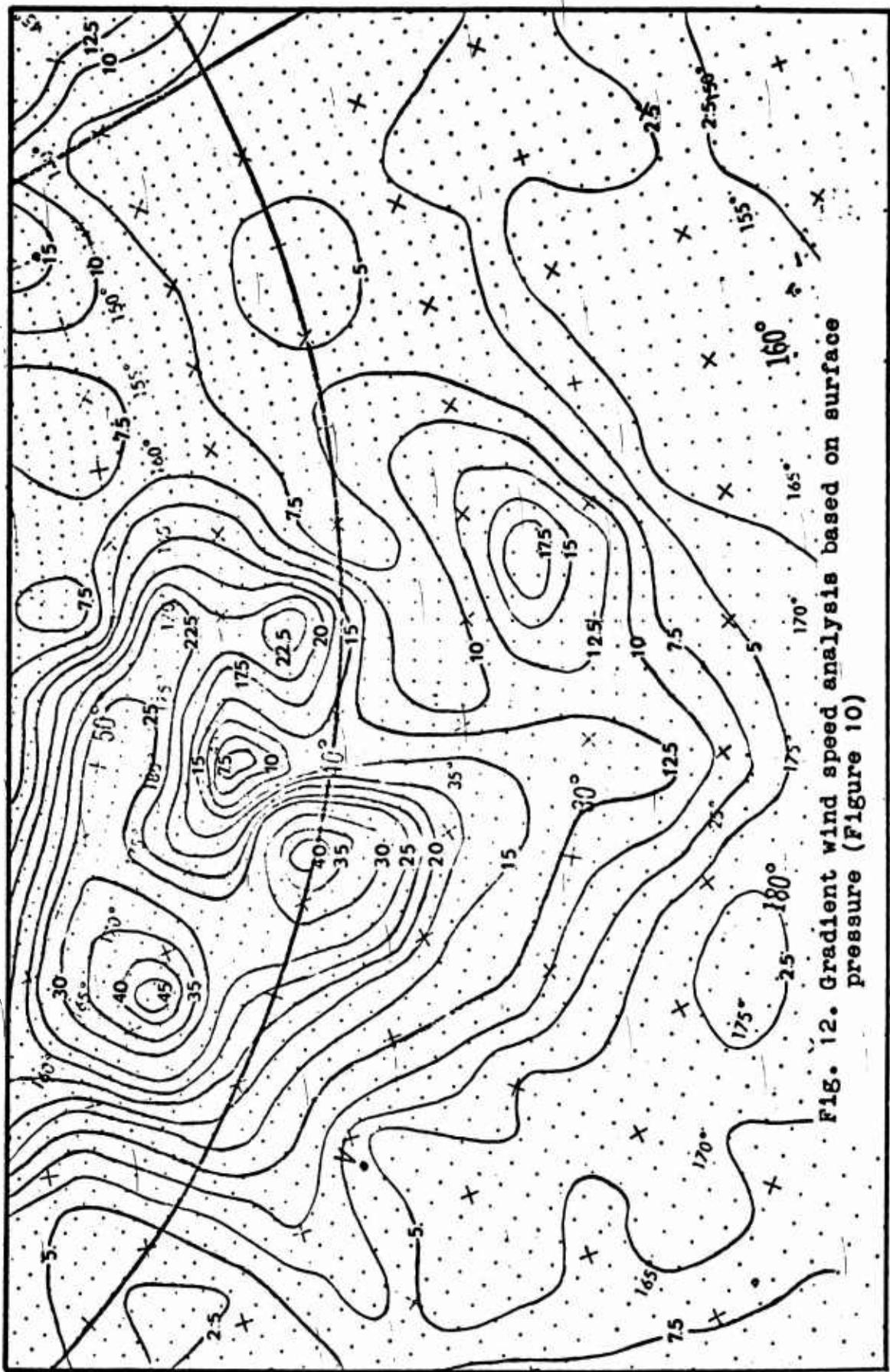


Fig. 12. Gradient wind speed analysis based on surface pressure (Figure 10)

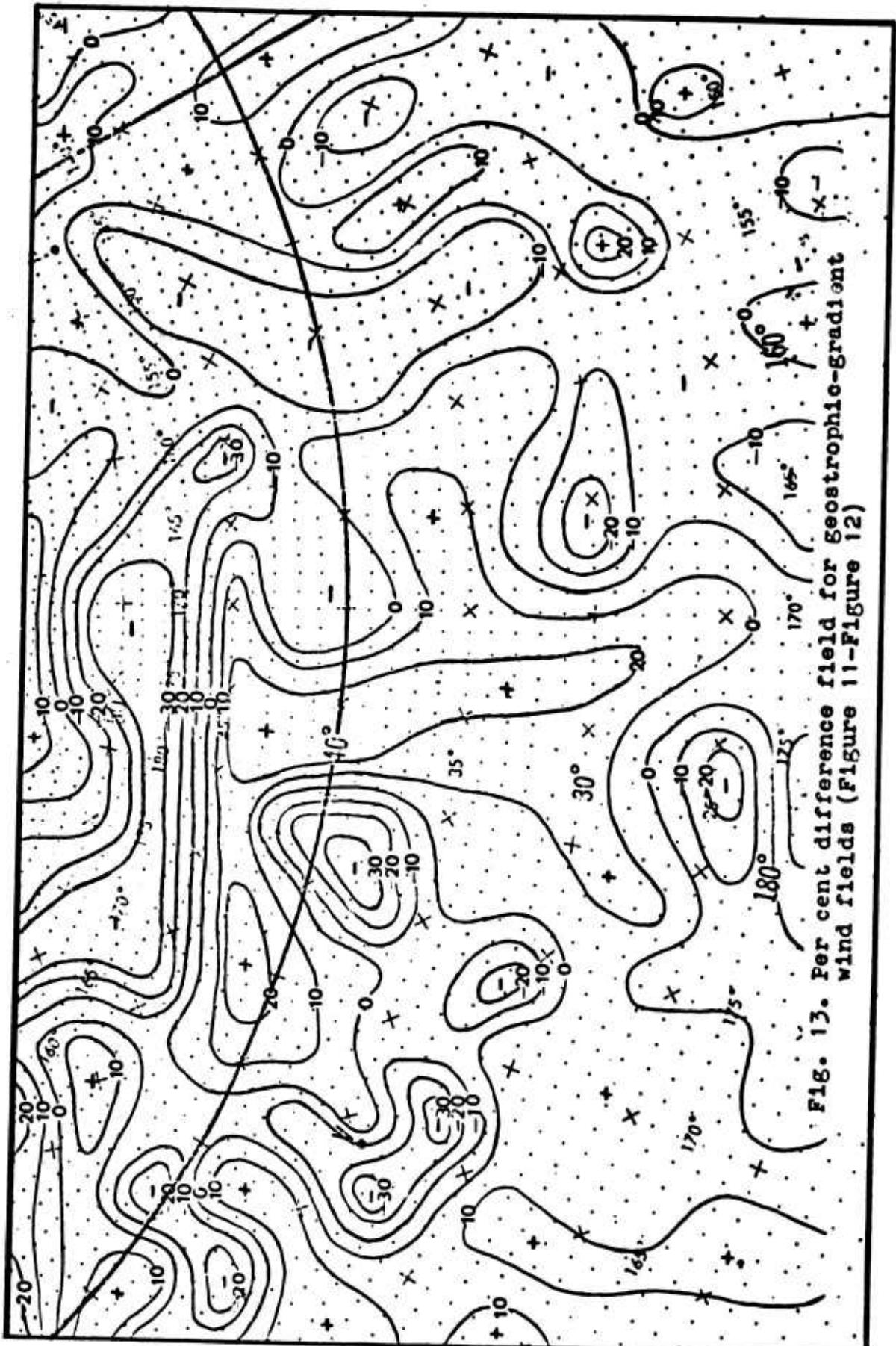


Fig. 13. Per cent difference field for geostrophic-gradient wind fields (Figure 11-Figure 12)

## B. SUBJECTIVE ANALYSIS

The results from the hand analyzed surface chart were also used in comparisons of results. Figure 14 illustrates the surface pressure and air-sea temperature difference fields from the subjective analysis which are analogous to Figure 10. The major variations between Figure 10 and 14 appear to be due to the difference in the number of synoptic reports used as well as the analysis scheme. Theoretically, Figure 14 should be more representative since more data were included, but it was a subjective analysis. Figures 15 and 16 depict the geostrophic and gradient surface winds computed from the subjectively analyzed pressure field.

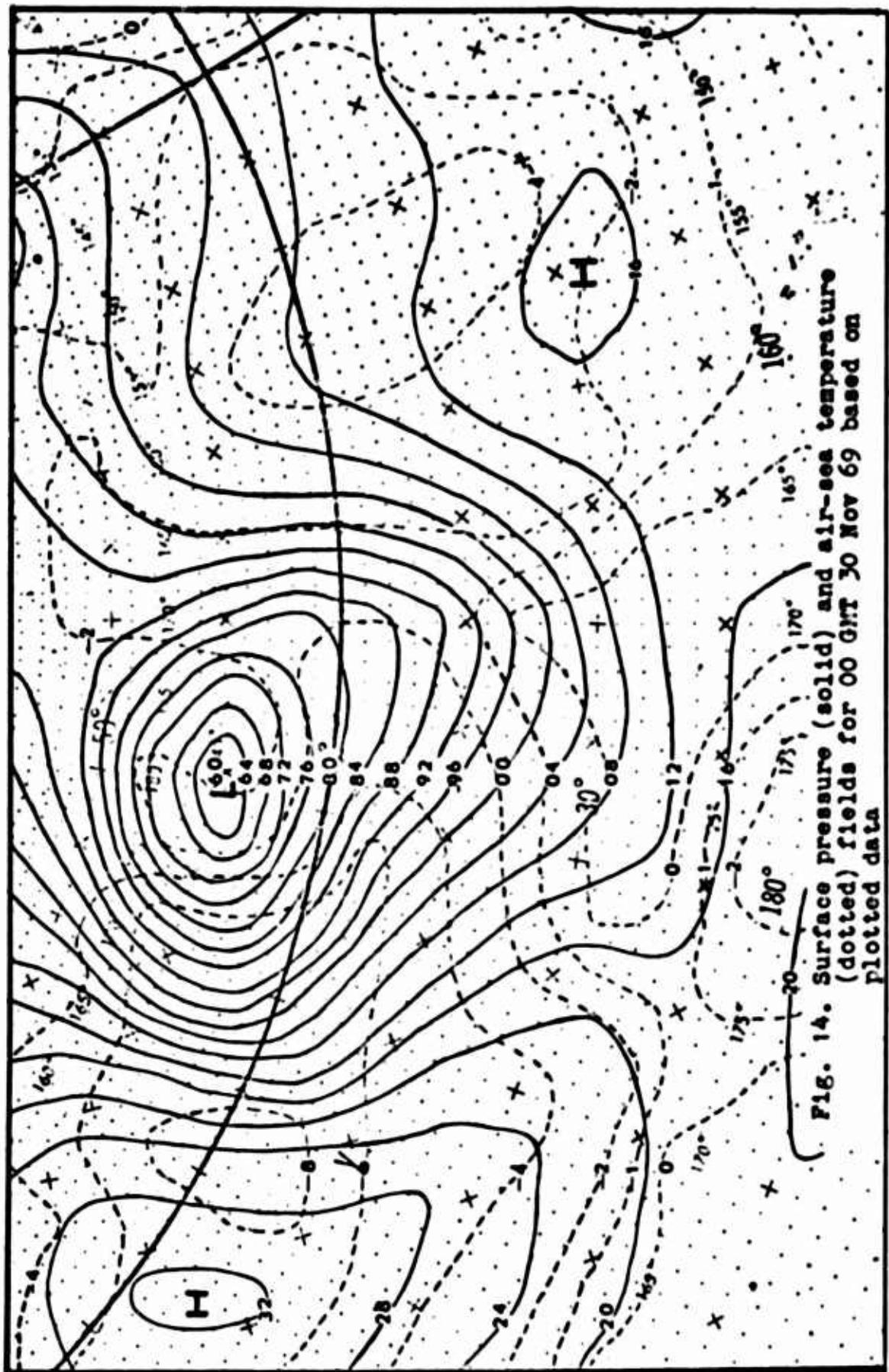


Fig. 14. Surface pressure (solid) and air-sea temperature (dotted) fields for 00 GMT 30 Nov 69 based on plotted data

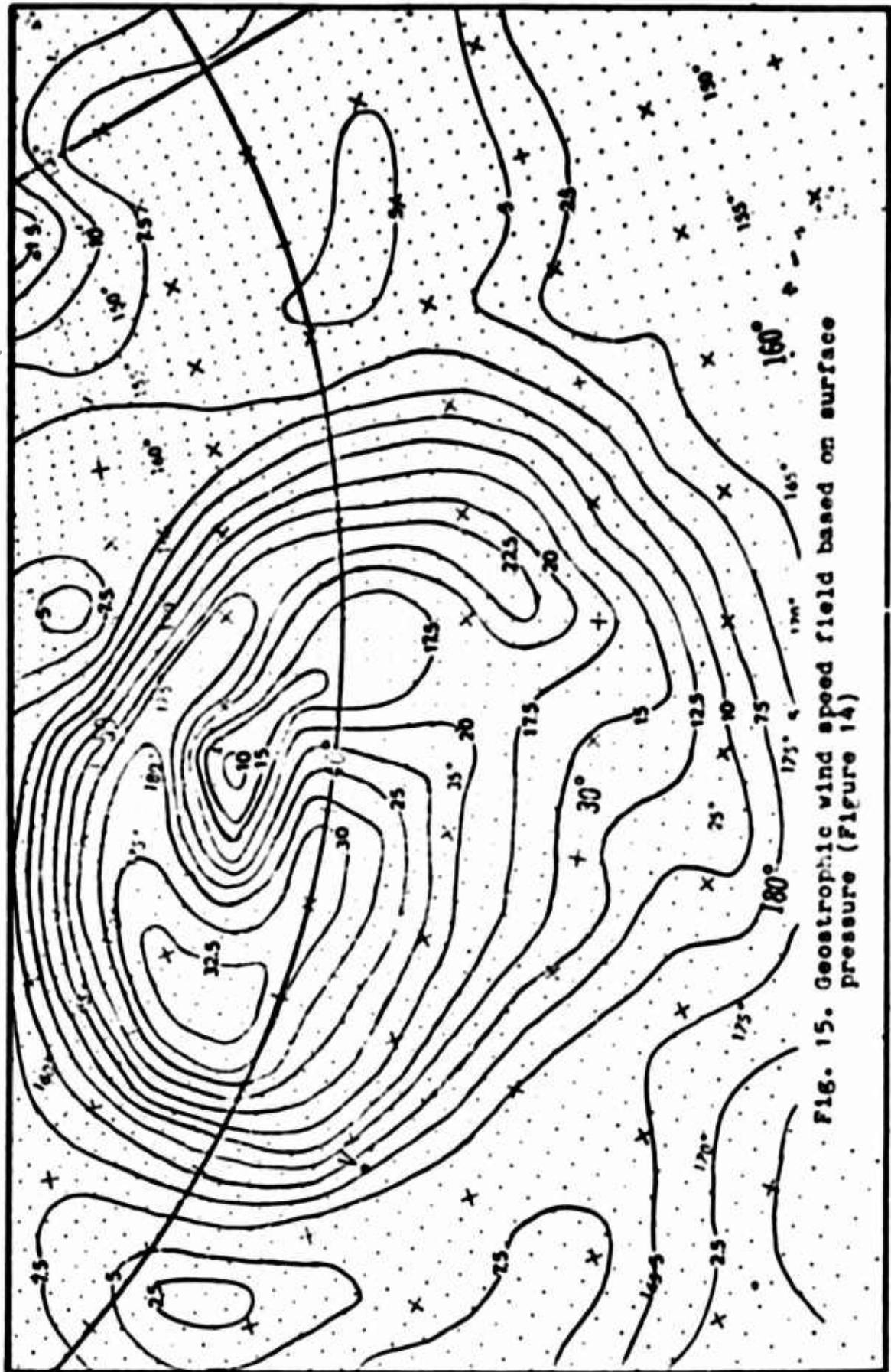


Fig. 15. Geostrophic wind speed field based on surface pressure (Figure 14)

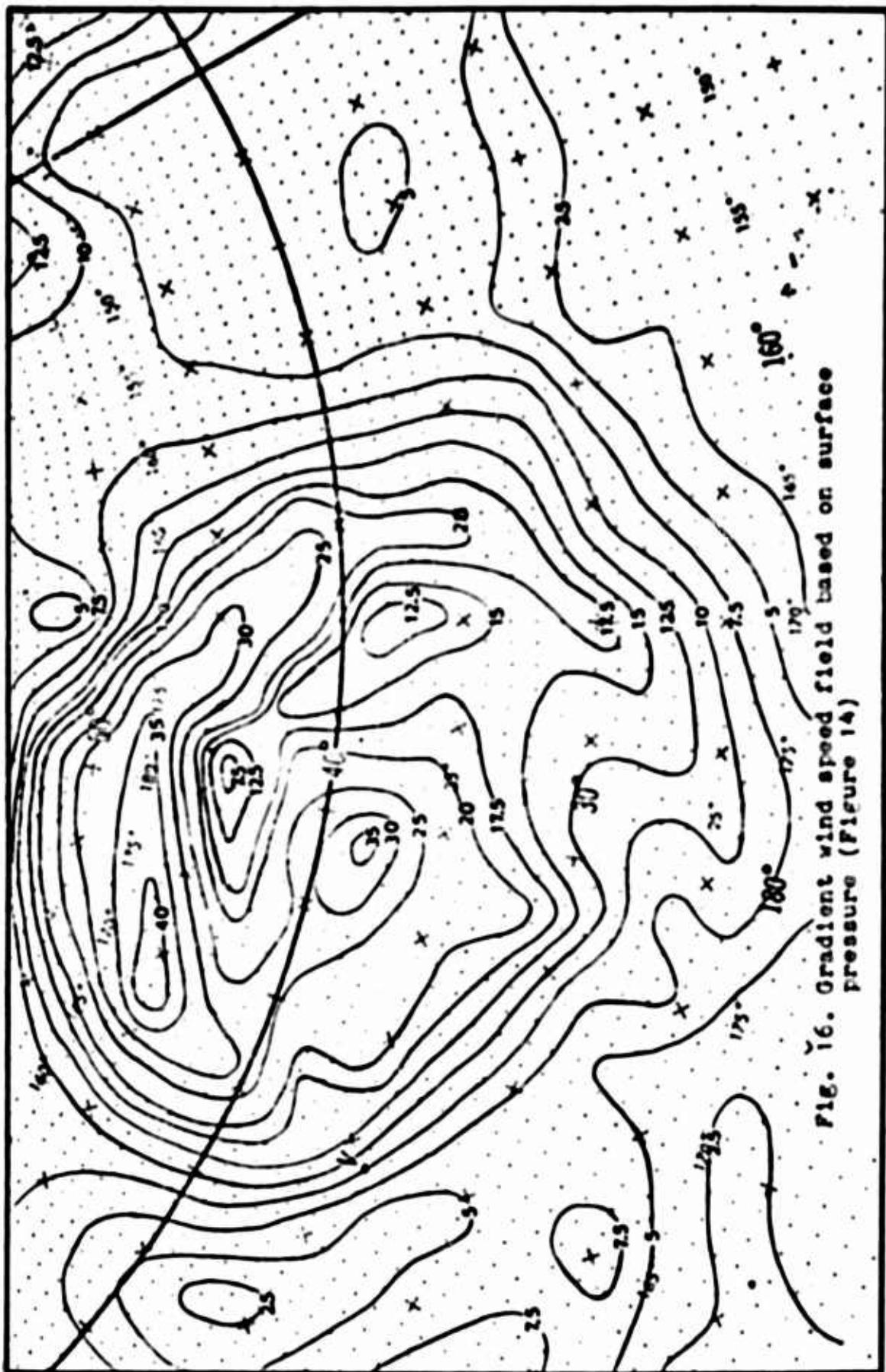


FIG. 16. Gradient wind speed field based on surface pressure (Figure 14)

## V. RESULTS FROM EXPERIMENTS WITH THE MODEL

### A. COMPARISONS IN TERMS OF SURFACE ROSSBY NUMBER AND STABILITY CATEGORIES

Near surface parameters were computed for the 19.5 meter level. This level represents an average height of ship anemometers (Bunting, 1968). Theoretically the computed 19.5 meter level winds could be compared with the observed wind fields and this would yield some information on the accuracy of the model in predicting surface winds.

In order to verify the computations and also to examine the range of conditions provided by the data, a comparison was made with Cardone's results which appear in Figures 4 and 5. Results for this comparison were obtained from computations in which stability was included but not baroclinicity. Results from all nine time periods appear in Figures 17 and 18 where values of the logarithm of the surface Rossby number (24) range from 7.76 to 9.45. The  $U_*/G$  versus  $\log R_o$  results (Figure 17) agree with Cardone's results (Figure 4) except with regards to the interval between  $L_*$  isolines. In Figure 17 this interval (spacing) is larger. With respect to  $\Psi_o$  versus  $\log R_o$  (Figure 18), the unstable results (negative  $L_*$ ) agree with those in Figure 5 except, again, with regards to the interval between  $L_*$  isolines. The stable results (positive  $L_*$ ), however, deviate considerably from Cardone's results with the veering angle decreasing as  $L_*$  increases. The results in Figure 18 would

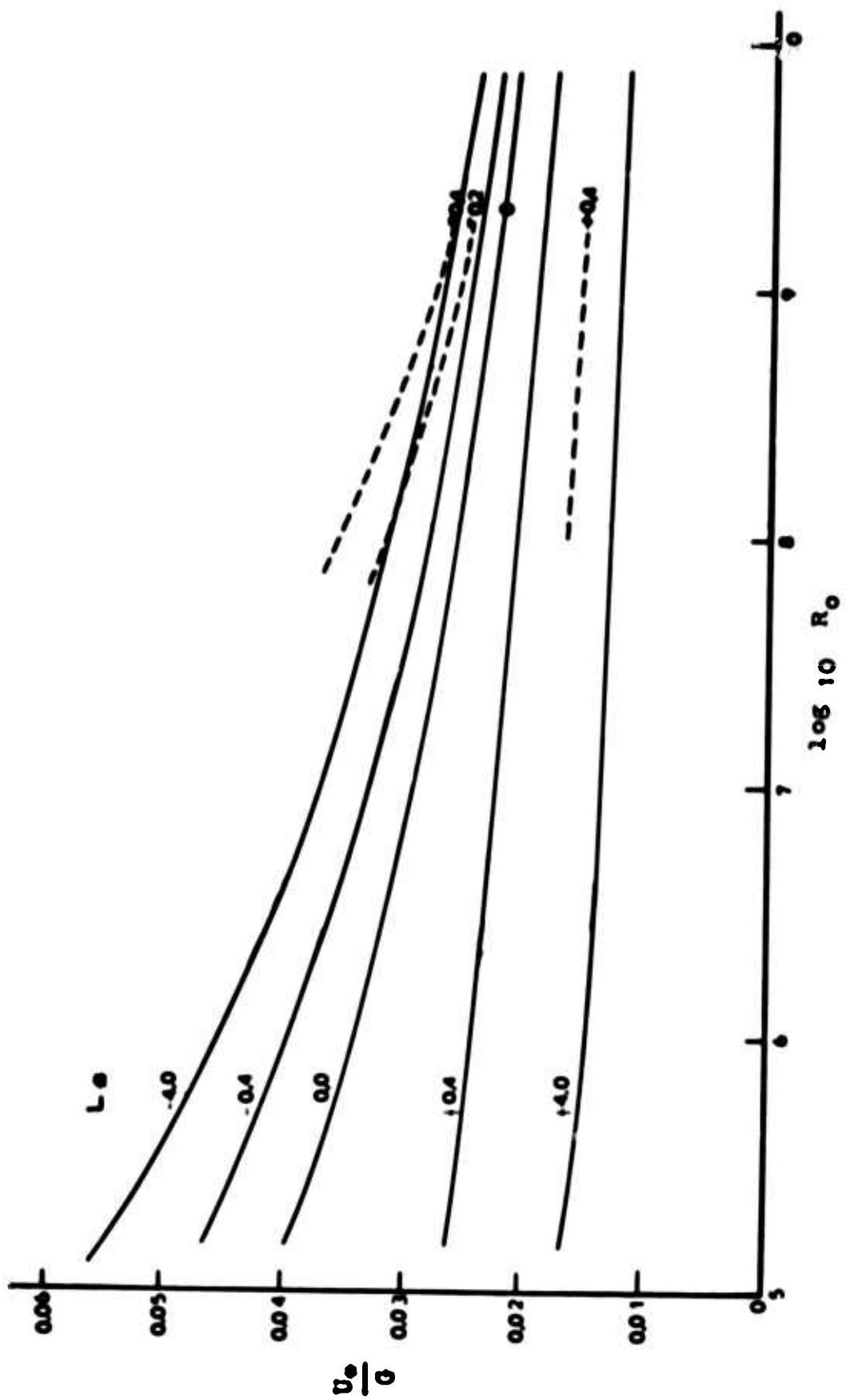


FIG. 17. Same as Figure 4 except dotted lines are results from this study

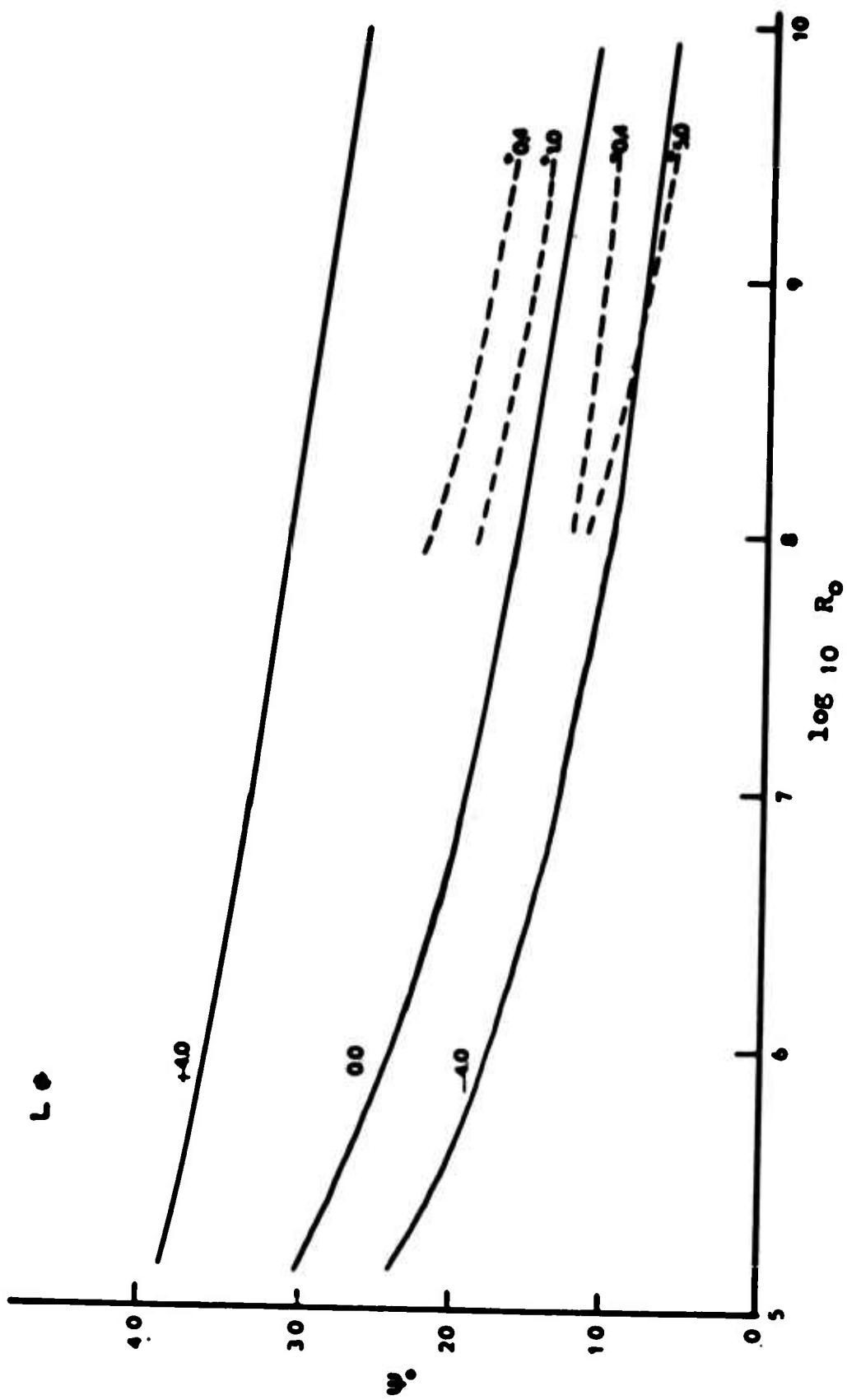


Fig. 18. Same as Figure 5 except dotted lines are results from this study

indicate that the veering angle decreases with both increasing stability and increasing instability with a discontinuity occurring near neutral conditions.

The unusual, and unexpected, features in the results just described were examined in considerable detail. The same results were noted by Pafias (unpublished study, 1971) in an investigation with Cardone's program. These features in the results are perhaps related to the drag coefficient representation selected by Cardone and also because different iteration limits were used in this study than those used by Cardone.

The range of data with respect to  $U_w/G$  and  $\theta_a - \theta_s$  was also examined and the results appear in Figure 19. The computations for these results were made on the basis of a barotropic atmosphere with no correction for isobaric curvature. These computations differ from those used by Cardone for Figure 8 in that a variable coriolis parameter was used. Isolines have been drawn for geostrophic wind speeds ranging from 15 to 75 knots. For those wind speeds considered by Cardone in Figure 19, and also for wind speeds above 20 knots considered by Pafias, these results are in agreement. The change in shape of the curves for winds under 20 knots was not described by Cardone but is probably due to a combination of the functional forms for the stability dependence and the drag coefficient as chosen by him.

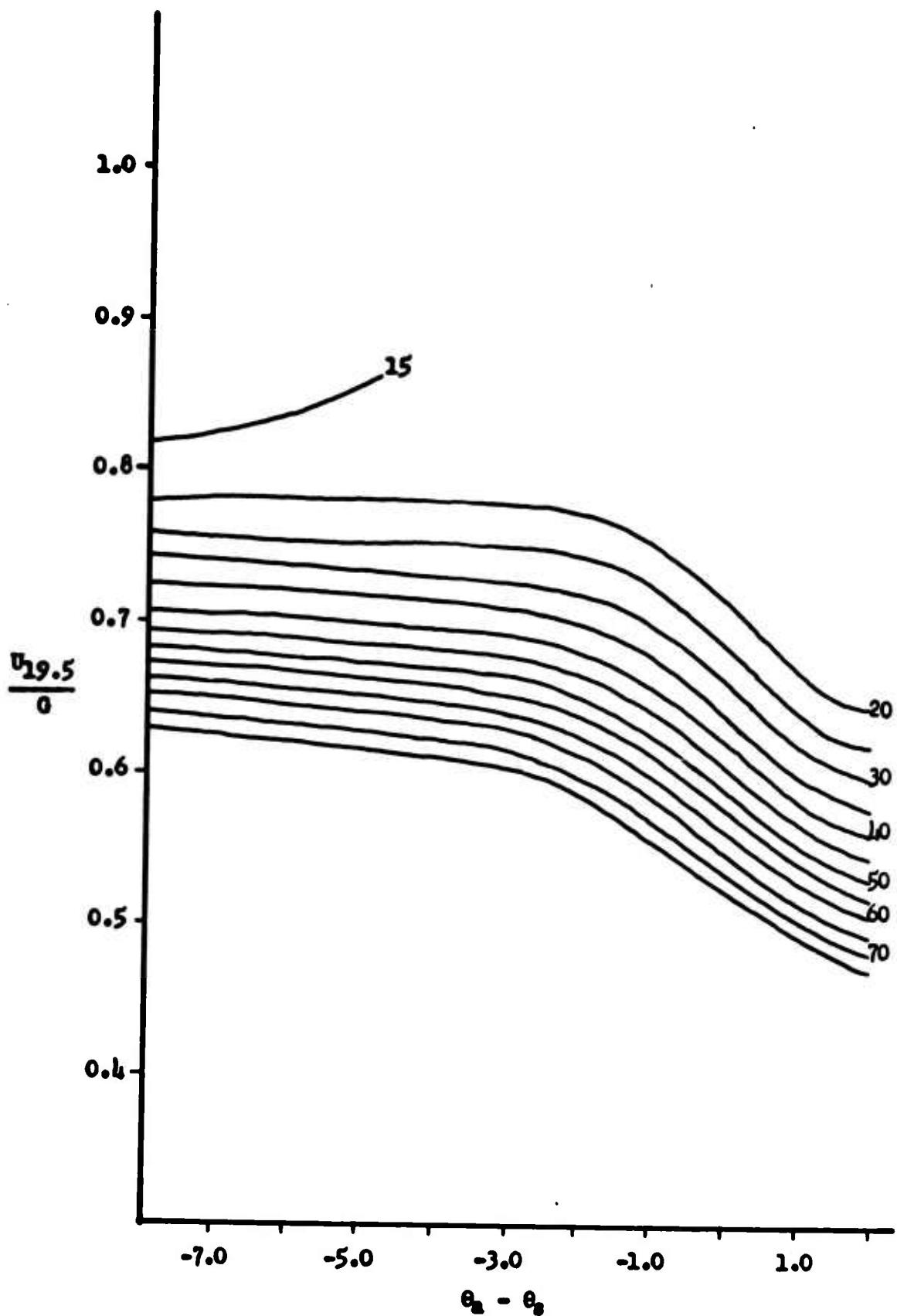


Fig. 19. Geostrophic drag coefficient versus air-sea temperature difference for range of geostrophic wind

## B. EXPERIMENTS WITH CHANGING THE PHYSICS IN THE TWO-LAYER MODEL AND CHANGING THE SPECIFICATION OF THE SURFACE GEOSTROPHIC WIND

In this section  $U_{19.5}$  results which were obtained from different computations with the two-layer model are compared. These computations were varied by including or not including specific parameters or corrections. The variations considered and compared were described earlier and are 1) N/NC/NT, 2) NN/NC/NT, 3) NN/NC/T, 4) NN/C/T, and 5) NN/NC/T with  $k = 0.35$ .

These variations will be referred to as Cases 1 through 5. It is noted that if the geostrophic wind speed was less than or equal to 5.0 meters/second, stability and baroclinicity were not taken into account in the iteration schemes. In regions with this wind speed range, especially in the subtropics, differences will not arise except where isobaric curvature exists.

Per cent difference (1-2) will be used to compare the two cases where (1-2) corresponds to the following computation at each grid point:

$$\frac{\text{case 1 result} - \text{case 2 result}}{\text{case 1 result}} \times 100 .$$

Although large differences between cases were unexpected, small corrections at low wind speeds could produce large per cent changes.

The surface wind fields corresponding to each of the five cases are presented in the following paragraphs along with selected comparisons between pairs of cases.

### Case 1 (N/NC/NT)

This is the simplest case with respect to the possible physics which are included in the model. The external wind field is defined for a neutral, barotropic atmosphere with no curvature correction. The wind field at 19.5 meters for this case appears in Figure 20.

### Case 2 (NN/NC/NT)

The effects of stability are now included and the 19.5 meter wind field for this case appears in Figure 21. The per cent difference (2-1) field is shown in Figure 22. Positive values correspond to increases in the computed  $U_{19.5}$  field due to including the stability effect. This increase is quite evident on the western side of the cyclone. The patterns are not expected to be coincident with the air-water temperature isolines in Figure 10 because the stability effect is dependent on both the temperature gradient and the wind speed. The 20% contours near the eastern border are attributed to small corrections at low wind speeds as discussed previously.

### Case 3 (NN/NC/T)

The effects of including the baroclinicity or thermal wind are now included and the 19.5 meter wind field for this case appears in Figure 23. The per cent difference (2-3) field is seen in Figure 24. Negative values correspond to an increase in the computed 19.5 meter wind due to including the thermal wind in the model. In this difference field, only 20 grid points (6%) appeared to be significantly affected by including the thermal wind. These grid points generally lie in an area of warm air advection.

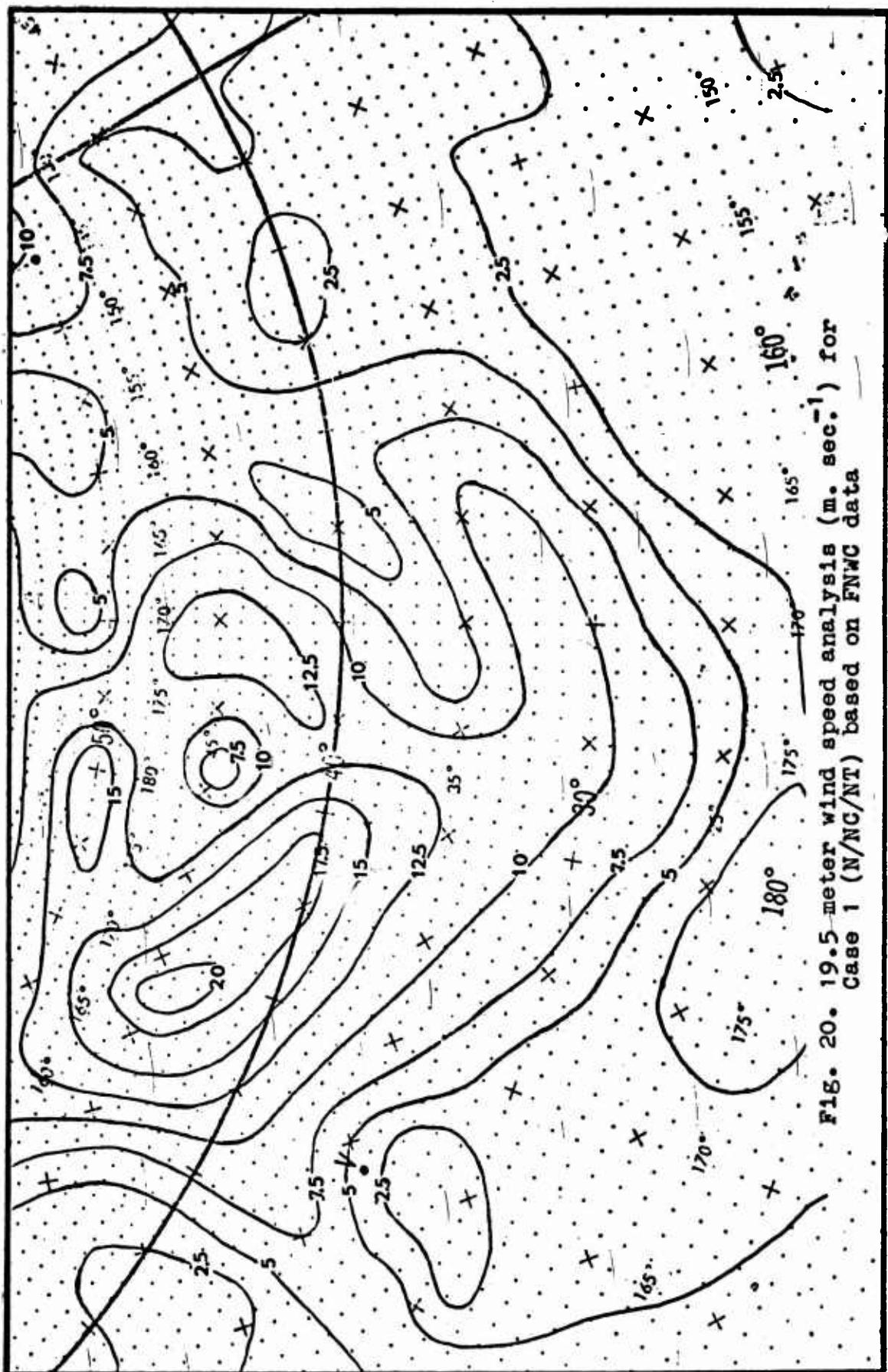


Fig. 20. 19.5-meter wind speed analysis (m. sec.<sup>-1</sup>) for Case 1 (N/NC/NT) based on FNNC data

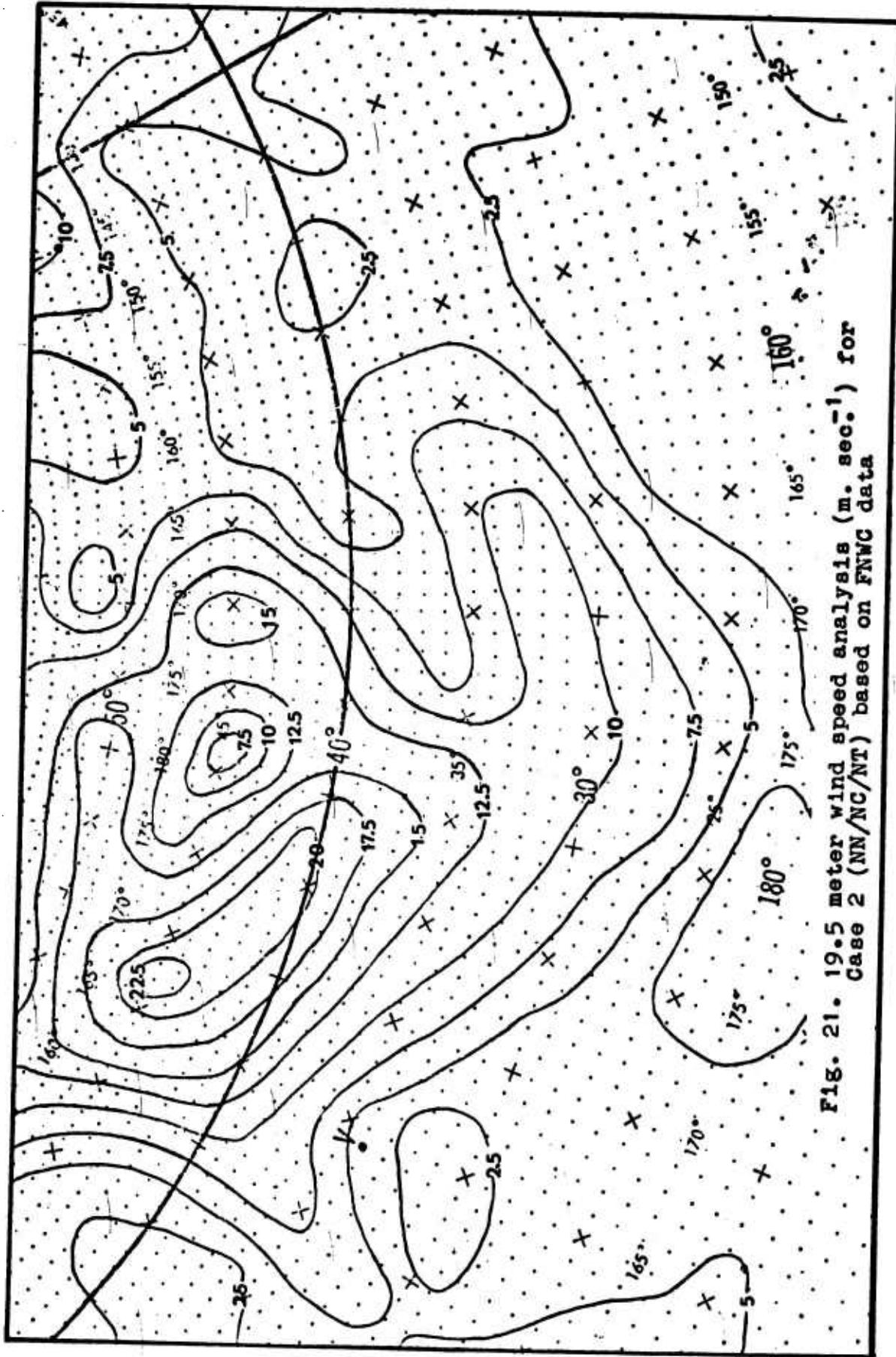


FIG. 21. 19.5 meter wind speed analysis ( $\text{m. sec}^{-1}$ ) for Case 2 (NN/NC/NT) based on FNNC data

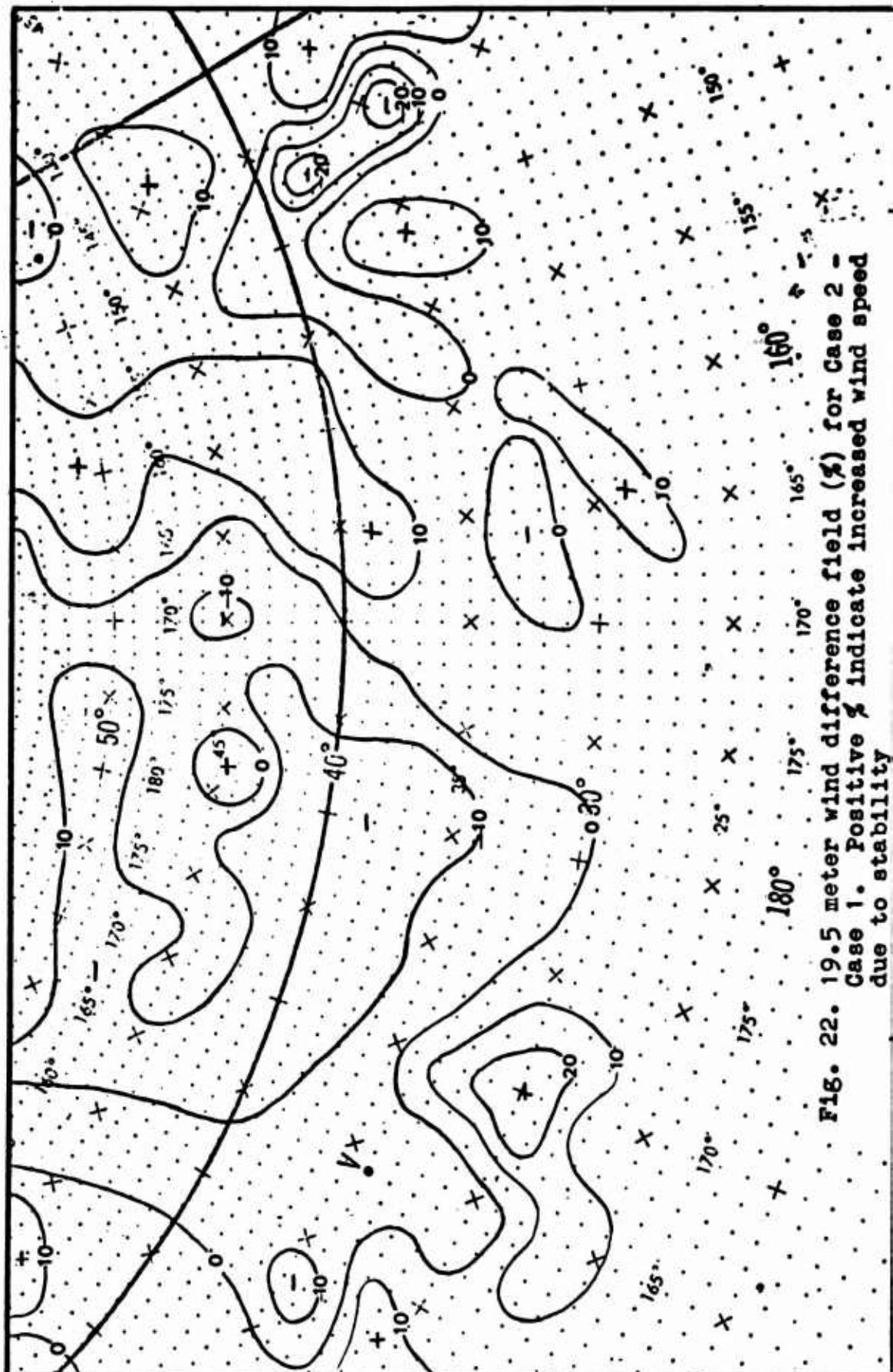


Fig. 22. 19.5 meter wind difference field (%) for Case 2 - Case 1. Positive % indicate increased wind speed due to stability

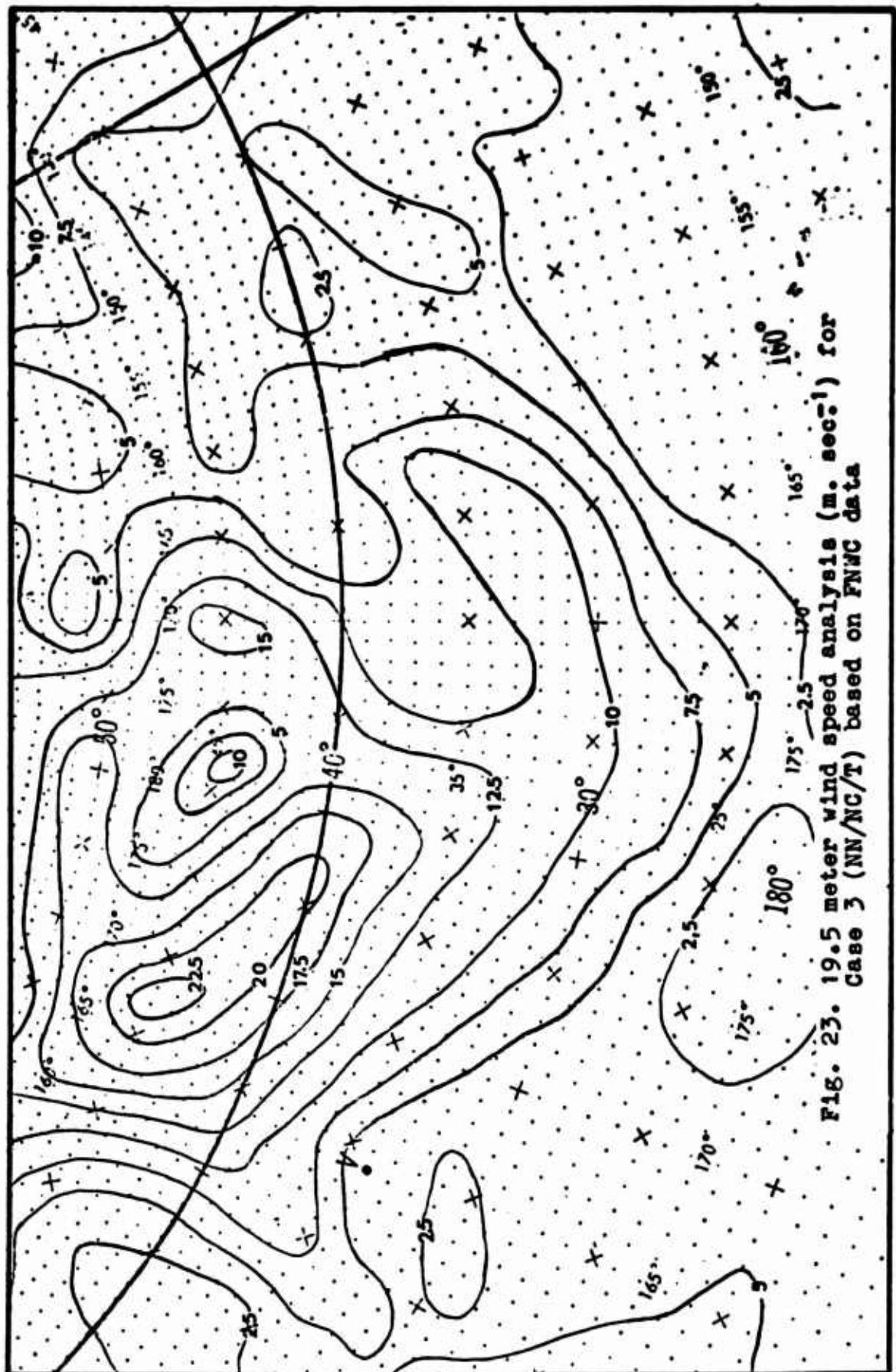


Fig. 23. 19.5 meter wind speed analysis (m. sec<sup>-1</sup>) for Case 3 (NN/NC/T) based on FNNC data.

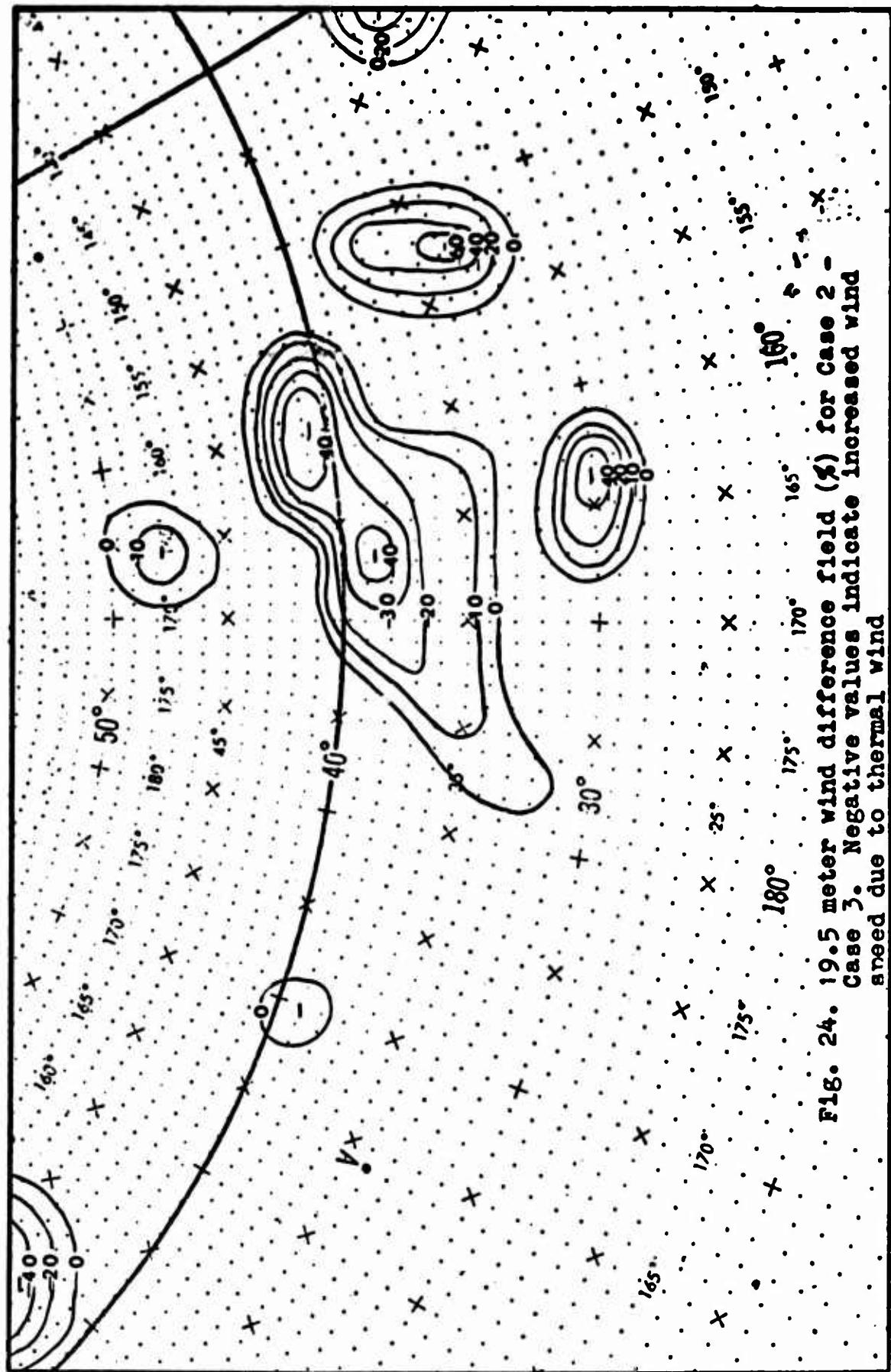


Fig. 24. 19.5 meter wind difference field (x) for Case 2 - Case 3. Negative values indicate increased wind speed due to thermal wind

#### Case 4 (NN/C/T)

In addition to all specifications possible within the physics of the model, the correction for curvature is applied to the external wind field. The 19.5 meter wind analysis for this field is shown in Figure 25 and the per cent difference field (3-4) appears in Figure 26. Negative values correspond to larger 19.5 meter winds due to including the curvature correction. Strong wind maximums, noted in comparing the external wind fields with and without this correction, also appear in Figure 26 as the predominant feature. In general, the effective change in the 19.5 meter wind field specification is larger and covers more area than any of the other cases.

#### Case 5 (NN/NC/T with $k = 0.35$ )

This case differs from case 3 only with respect to the change of the von Kármán constant from 0.40 to 0.35. The 19.5 meter wind field for this case appears in Figure 27 and the per cent difference field (3-5) is depicted in Figure 28. Negative values correspond to increased 19.5 meter winds due to the change in the value of  $k$ . The suggested value of  $k = 0.35$  arose from recent surface layer observational investigations by Businger et al. (1971). They noted from their results that using  $k = 0.40$  in several empirical expressions led to estimates of  $U_*$  values which were 15% too high when compared with direct measurements of the momentum transfer. The fields in Figure 28 indicate that a general increase of 12 to 17% occurs in the computed 19.5 meter wind field if the lower value of  $k$  is used. This increase agrees with the  $U_*$

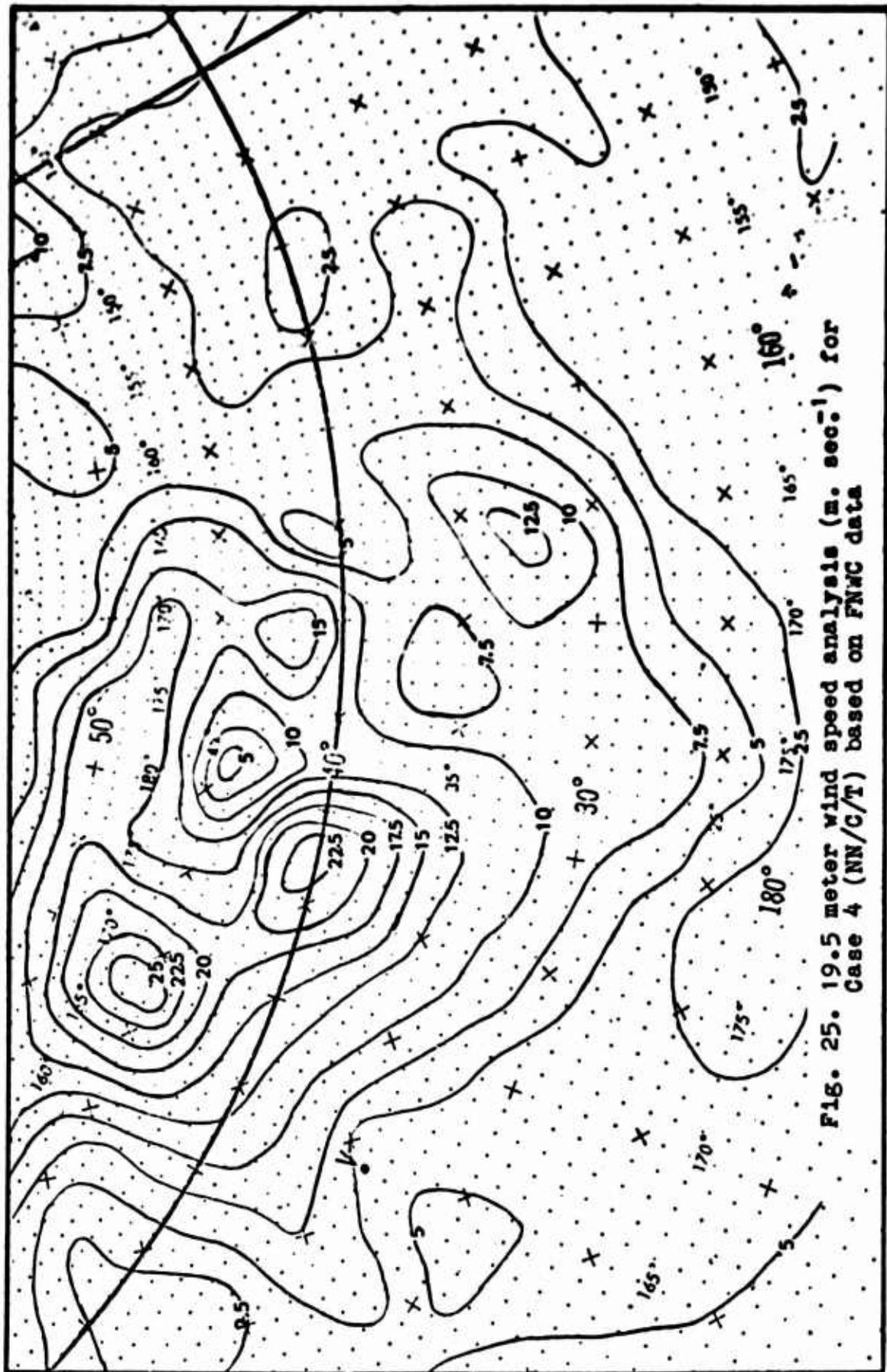


FIG. 25. 19.5 meter wind speed analysis ( $m \text{ sec}^{-1}$ ) for Case 4 (NN/C/T) based on FNWC data

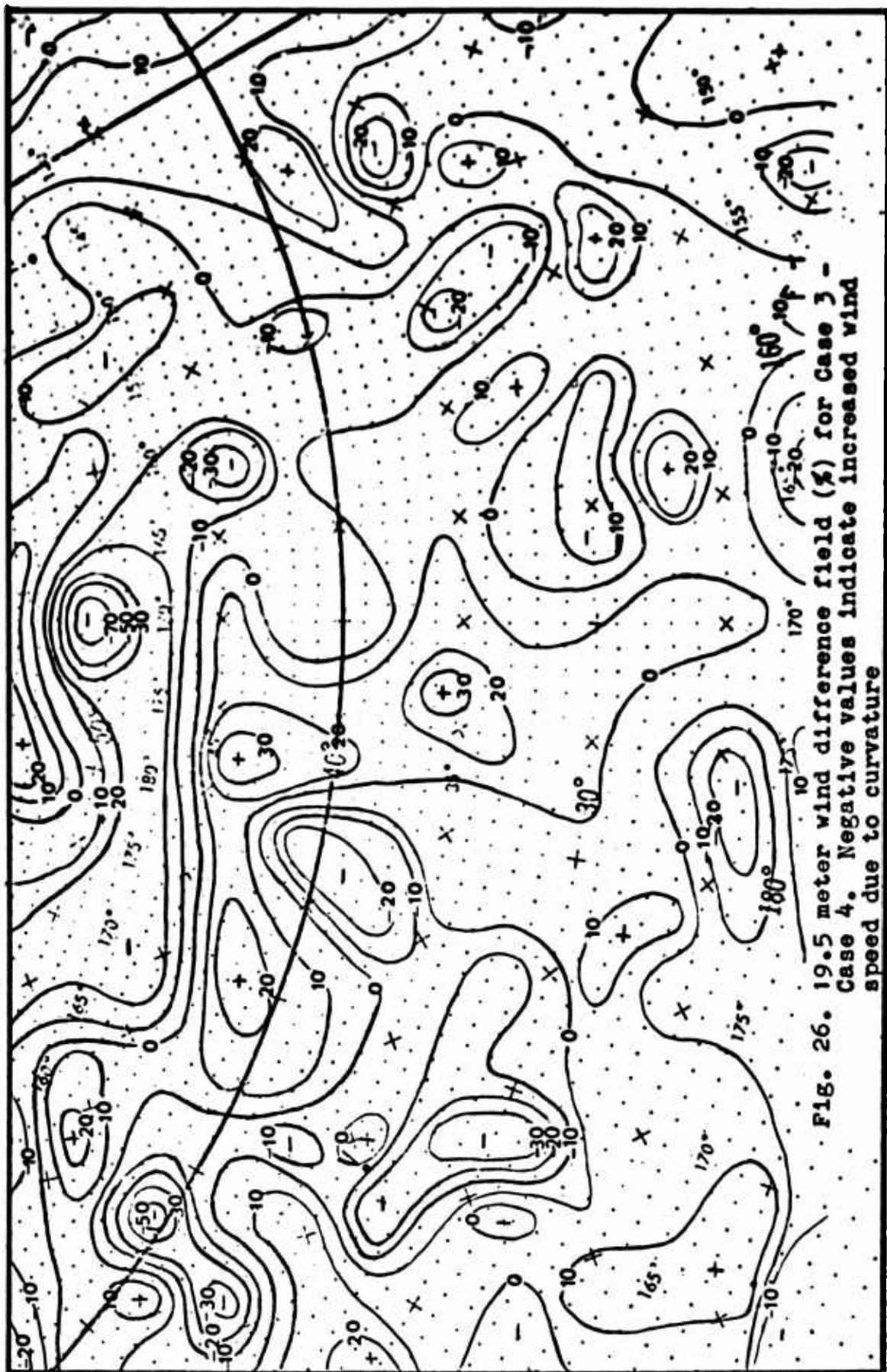


Fig. 26. 19.5 meter wind difference field (%) for Case 3 - Case 4. Negative values indicate increased wind speed due to curvature

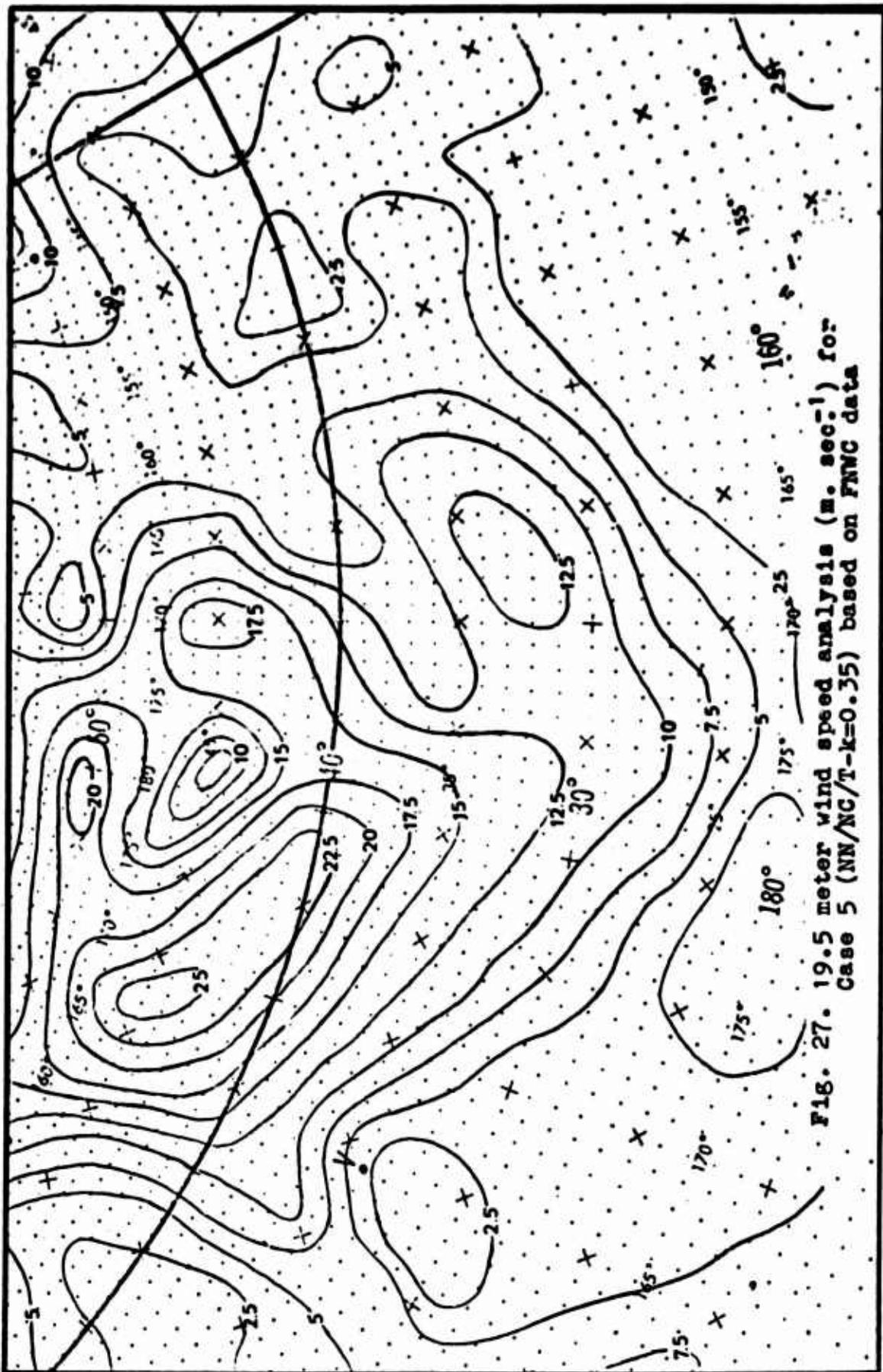


Fig. 27. 19.5 meter wind speed analysis ( $m. sec^{-1}$ ) for Case 5 ( $NN/NC/T-k=0.35$ ) based on FNNC data

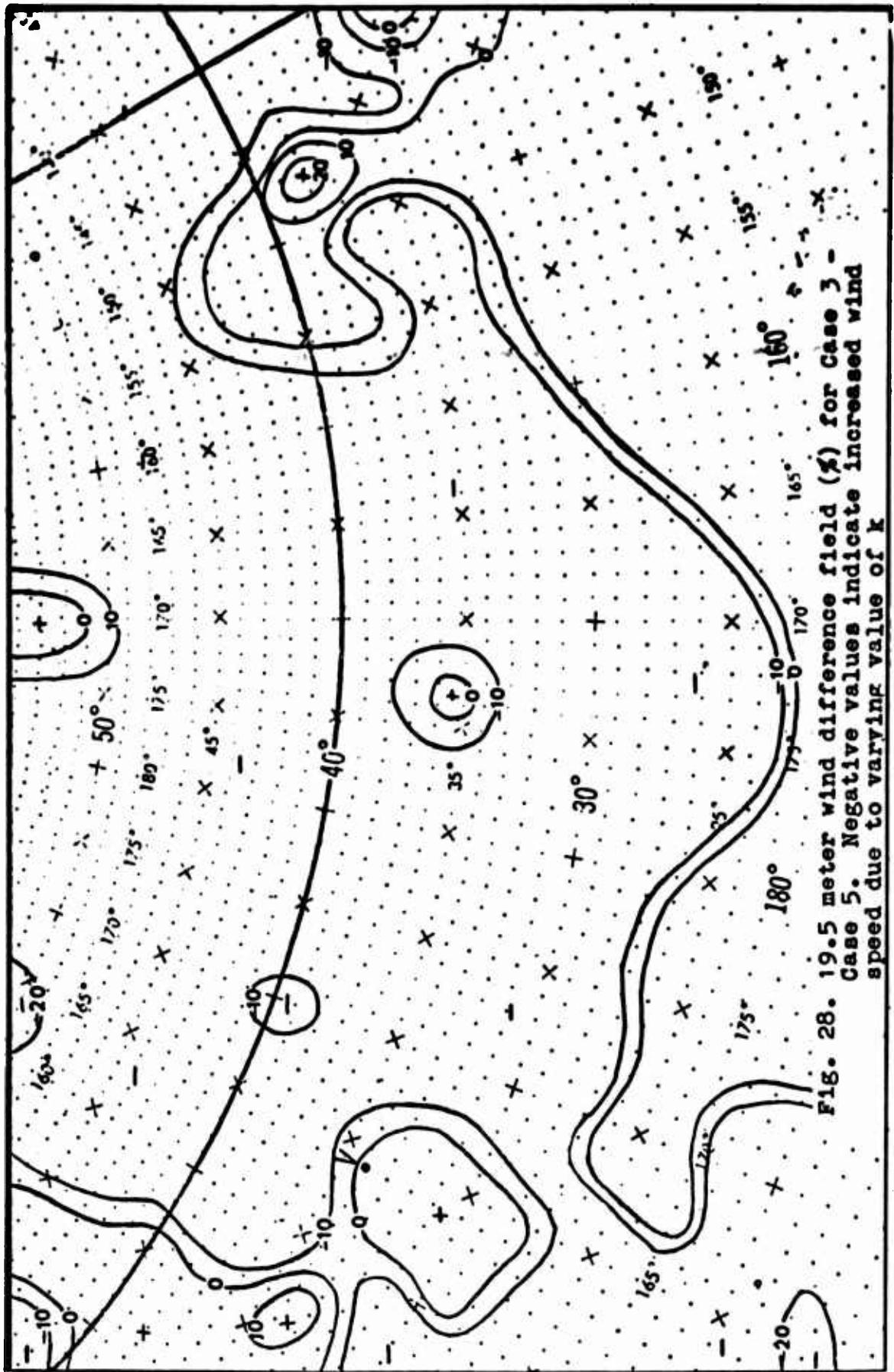


Fig. 28. 19.5 meter wind difference field (x) for Case 3 - Case 5. Negative values indicate increased wind speed due to varying value of  $k$

discrepancies noted by Businger et al.. Because the von Kármán constant appears in several expressions in the boundary layer model, the importance of its change in the two-layer model was not known. These results indicate that the difference between  $k = 0.35$  and  $k = 0.40$  is important.

In summary, the five cases described above indicate which terms were most influential in the model. It should be remembered that most of the subtropical points were not influenced by the model since the wind speeds were generally less than 5.0 meters/second. Of all the modifications to the model, the curvature correction to the external field produced the largest variation over the entire grid. Changing  $k$  from 0.40 to 0.35 accounted for the second most significant change. Differences in the surface wind fields were evident, however, when specifications in the model were made. Of the two possible specifications, stability appeared to be more significant than the thermal wind with respect to area influenced and per cent difference changes.

Although the influence of the thermal wind in this model appears to be small with respect to the relative magnitude of the computed wind, the influence it has on the direction of the wind appears to be significant. The difference field, in degrees, of the wind direction for NN/NC/T - NN/NC/NT appears in Figure 29. This field represents the difference in wind direction between a case with and a case without the thermal wind specification. The region of significant differences does not coincide with that region where the thermal wind effect was large

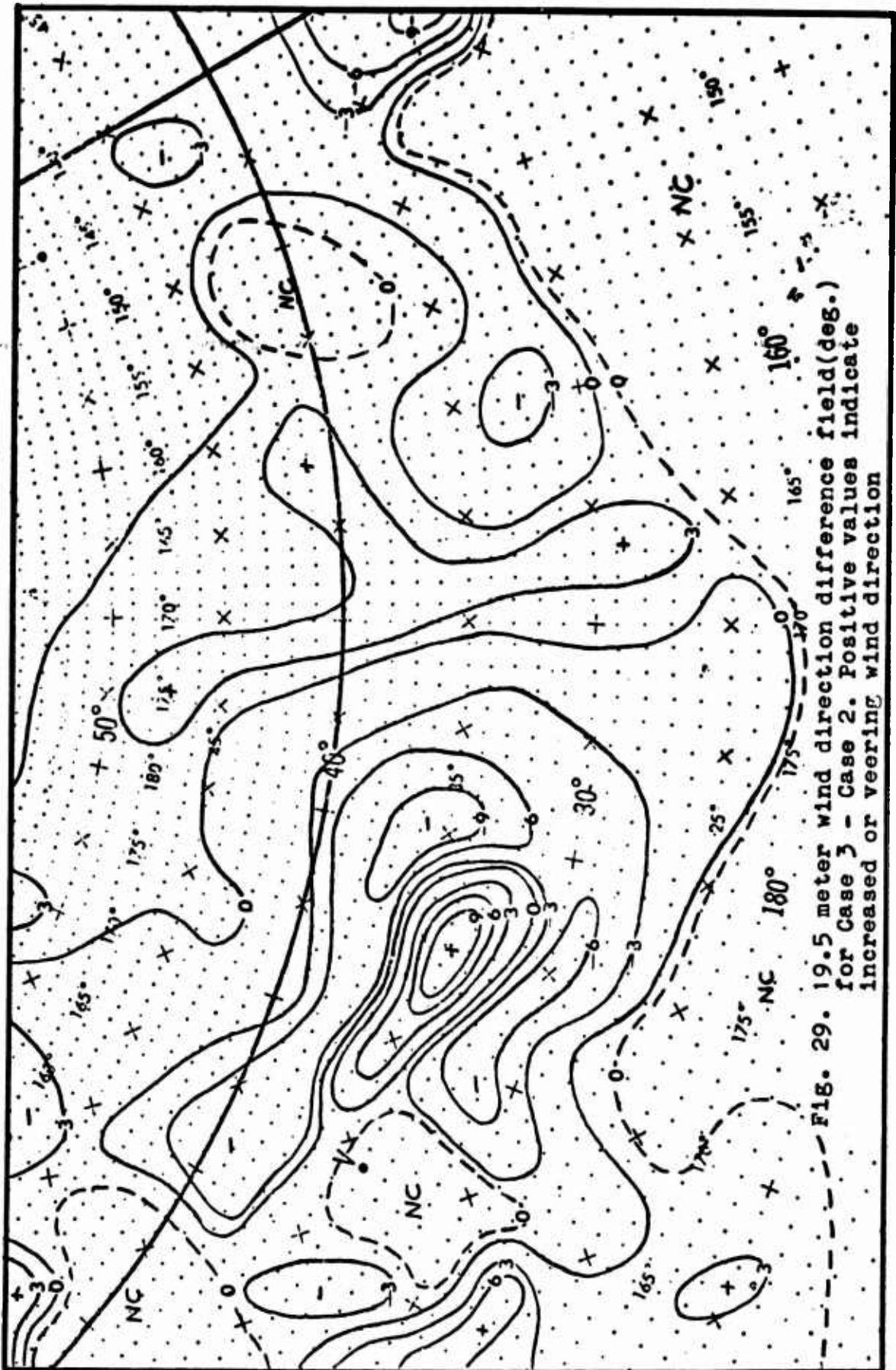


Fig. 29. 19.5 meter wind direction difference field(deg.) for Case 3 - Case 2. Positive values indicate increased or veering wind direction

with respect to wind speed. Positive values in Figure 29 correspond to a veering wind and negative values correspond to a backing wind. It is noted that the direction change due to baroclinicity is primarily determined by the angle between the thermal and geostrophic winds.

### C. COMPARISONS OF COMPUTED WINDS WITH OBSERVED WINDS

These comparisons will be described for the case NN/C/T with  $k = 0.35$ . The 19.5 meter wind fields computed from the other cases were compared to the subjectively analyzed observed surface wind field (Figure 30) and in all cases the computed winds underestimate the observed winds. This was also the case when the above corrections were included, but yielded the best results with respect to the magnitude and patterns of the wind speeds.

Because the previous comparisons revealed the importance of the specifications of the external wind field with respect to curvature, it was considered necessary in this comparison to examine the importance of the initial analysis (FNWC or subjective). The 19.5 meter wind field computed with the above specifications and with the FNWC analysis is shown in Figure 31 and the wind field based on subjective analysis is seen in Figure 32. The observed surface wind field which appears in Figure 30 was analyzed using only those observations reported above 10 meters/second. In Figures 31 and 32, the maximum contour, 30 meters/second, agrees with the observed maximum wind speed, but the objective analysis does not indicate the 25 meter/second maximum east of the low.

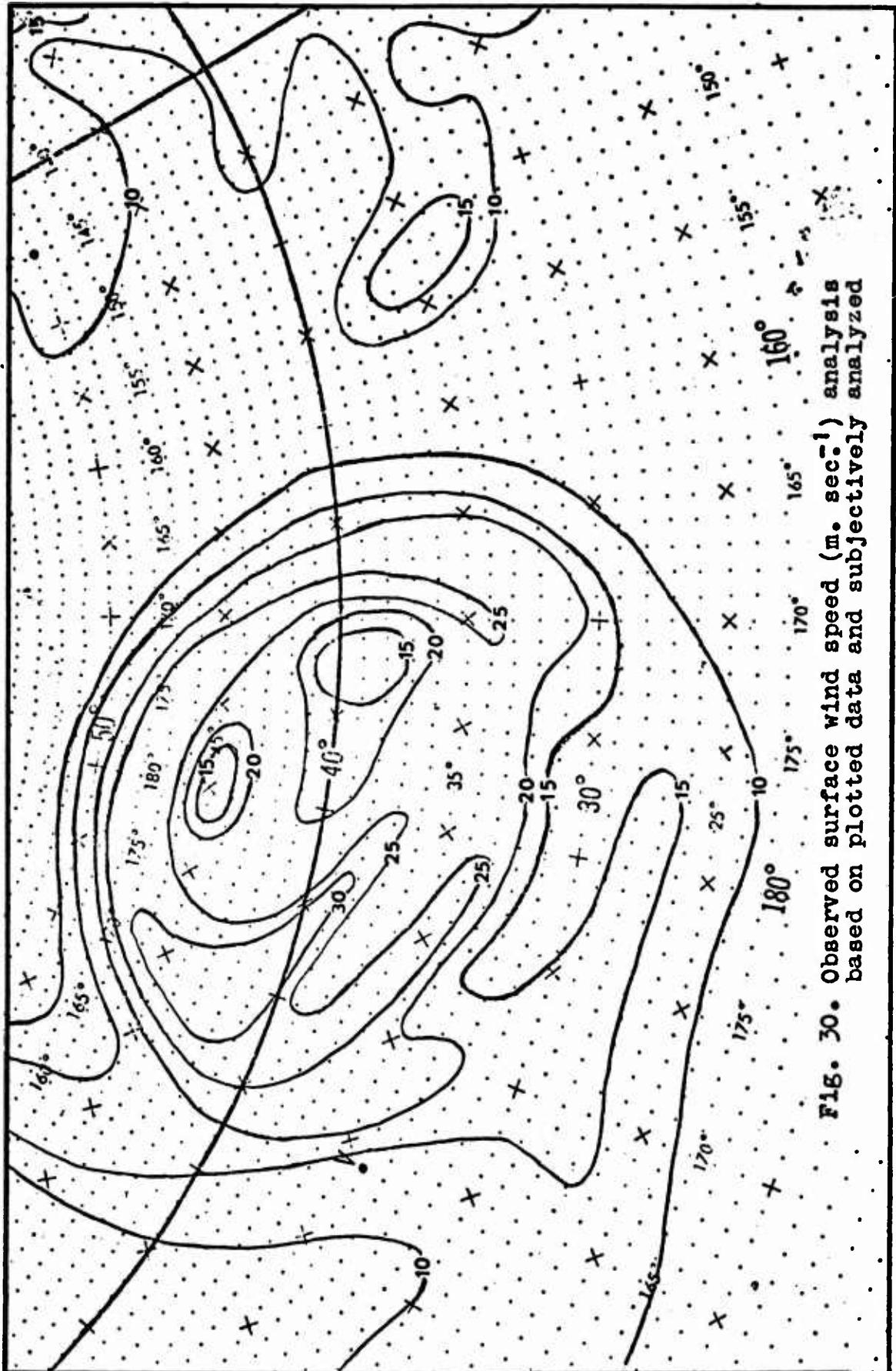


FIG. 30. Observed surface wind speed ( $m. sec^{-1}$ ) analysis based on plotted data and subjectively analyzed

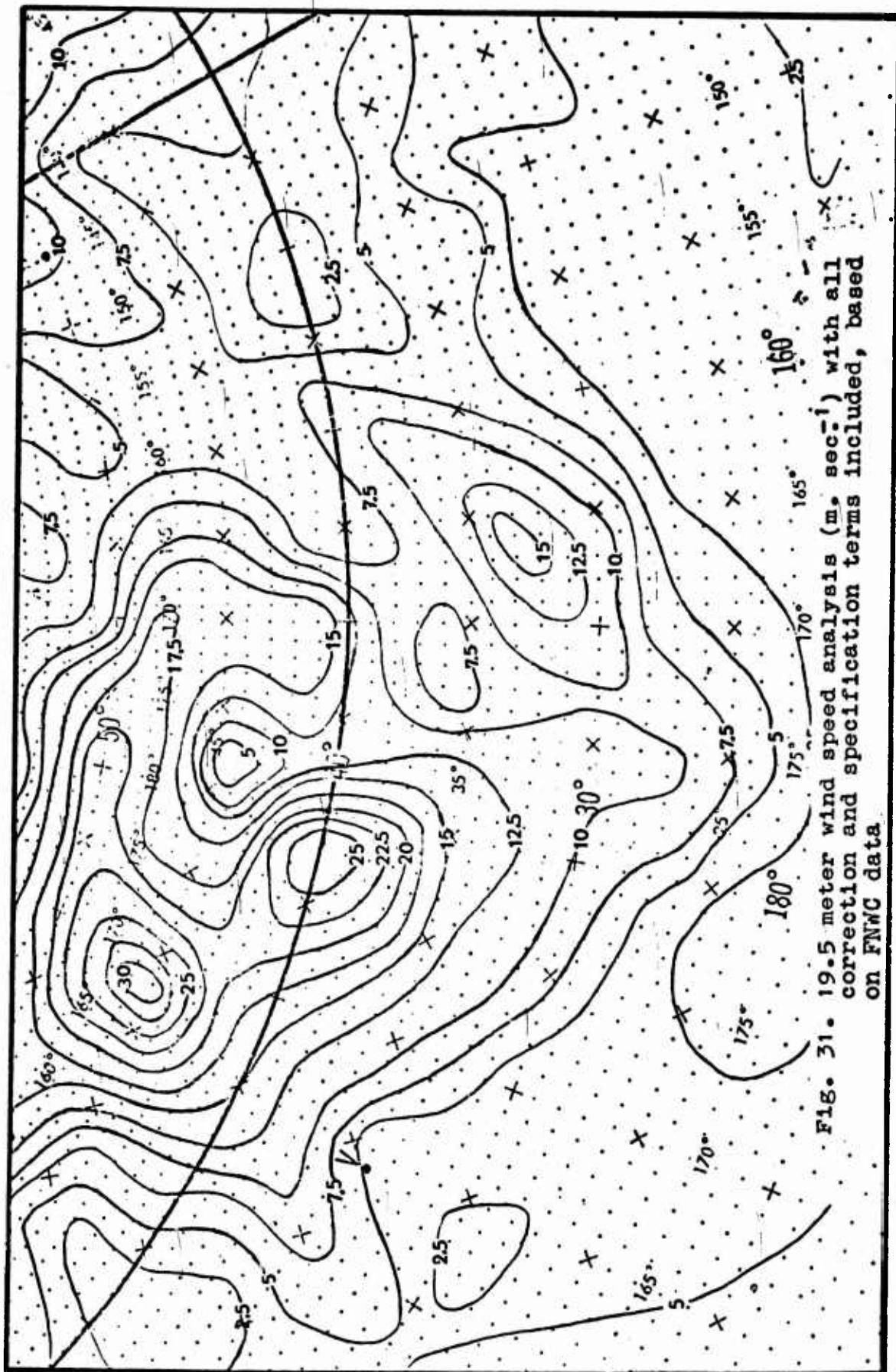


Fig. 31. 19.5 meter wind speed analysis ( $m, sec^{-1}$ ) with all correction and specification terms included, based on FNNG data.

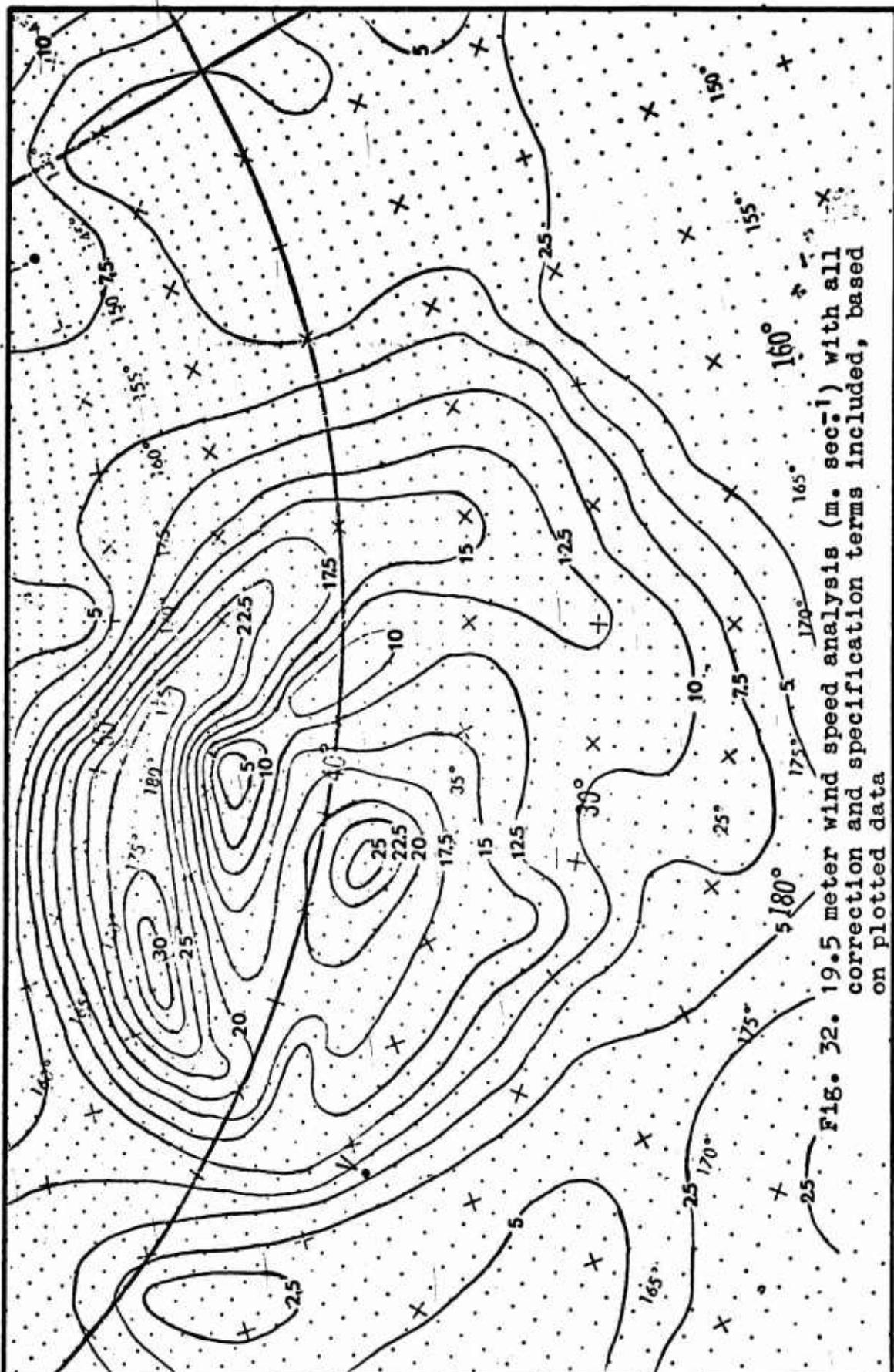


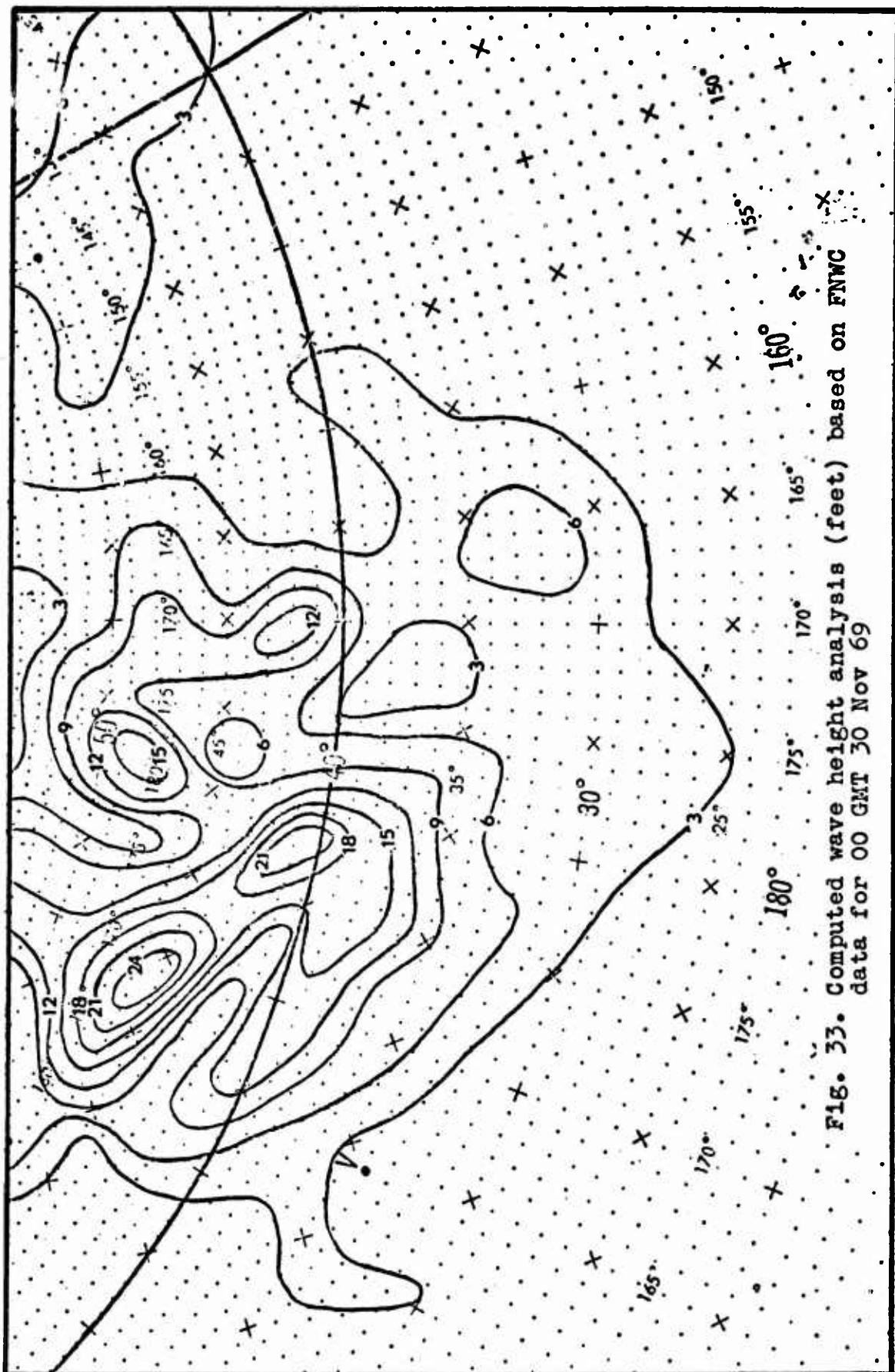
FIG. 32. 19.5 meter wind speed analysis (m. sec.<sup>-1</sup>) with all correction and specification terms included, based on plotted data

A comparison of computed 19.5 meter winds from the two analyses (Figures 31 and 32) and the external wind fields (Figures 11 and 15) shows that differences in the external wind fields are consistently carried through the model. The model, however, was able to produce 19.5 meter wind fields in both cases which were in good agreement with the observed wind field, at least for wind speeds over 10 meters/second.

#### D. RESULTS FROM WAVE HEIGHT COMPUTATIONS

Wave height fields were computed using the 19.5 meter wind fields obtained from the boundary layer model and the simplified wave height program developed from the FNWC operational sea-swell model. The program was run starting at 00 GMT 28 Nov in order to build history for this analysis. The 19.5 meter wind computations were obtained using all specifications and corrections described previously, and they were also made on both the FNWC and subjectively analyzed fields. The computed wave field from the FNWC data is seen in Figure 33 and the wave field from the subjective analysis appears in Figure 34. The observed wave field obtained from synoptic observations is depicted in Figure 35. It is noted that wave height is probably the most difficult parameter to observe and therefore a representative analysis of the observed wave field is difficult to achieve.

The wave field obtained from the subjective analysis, Figure 34, appears to be in good agreement with the observed heights with maximum computed heights slightly less than the maximum observed heights.



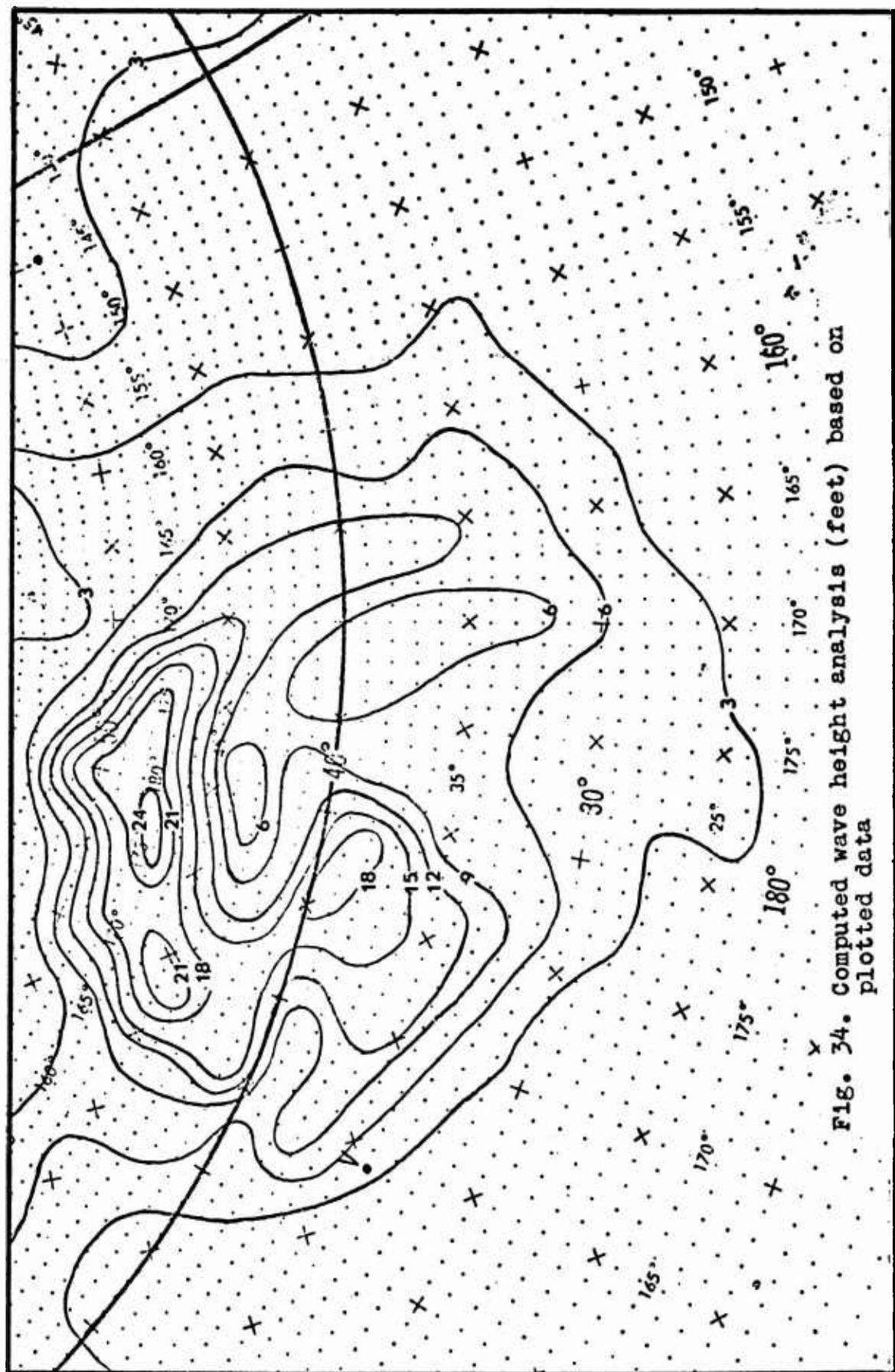


Fig. 34. Computed wave height analysis (feet) based on plotted data

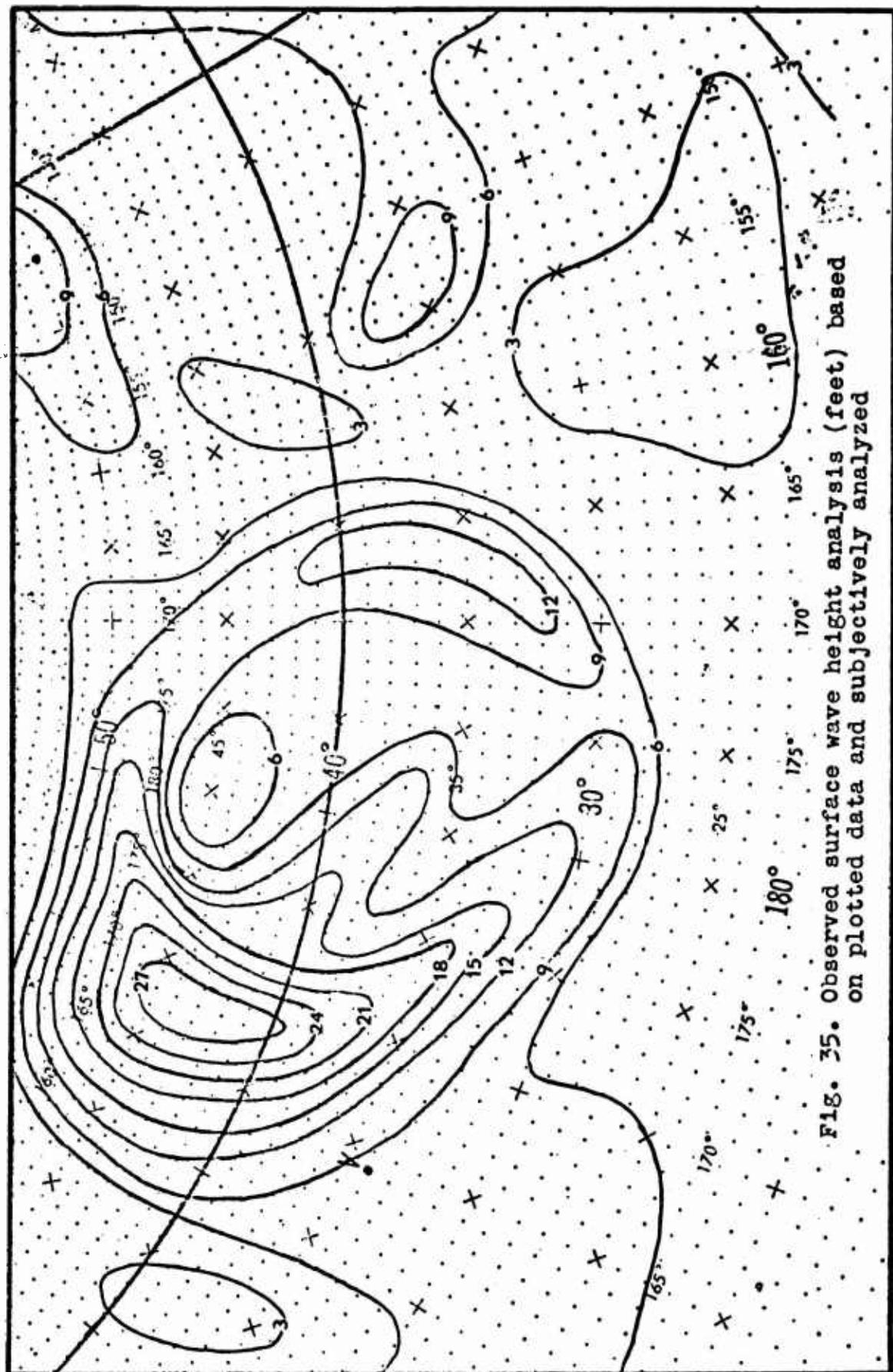


Fig. 35. Observed surface wave height analysis (feet) based on plotted data and subjectively analyzed

Results computed from the FNWC analysis, Figure 33, appear to have similar agreement with respect to the maximum height but poorer agreement with respect to the overall pattern of the wave field.

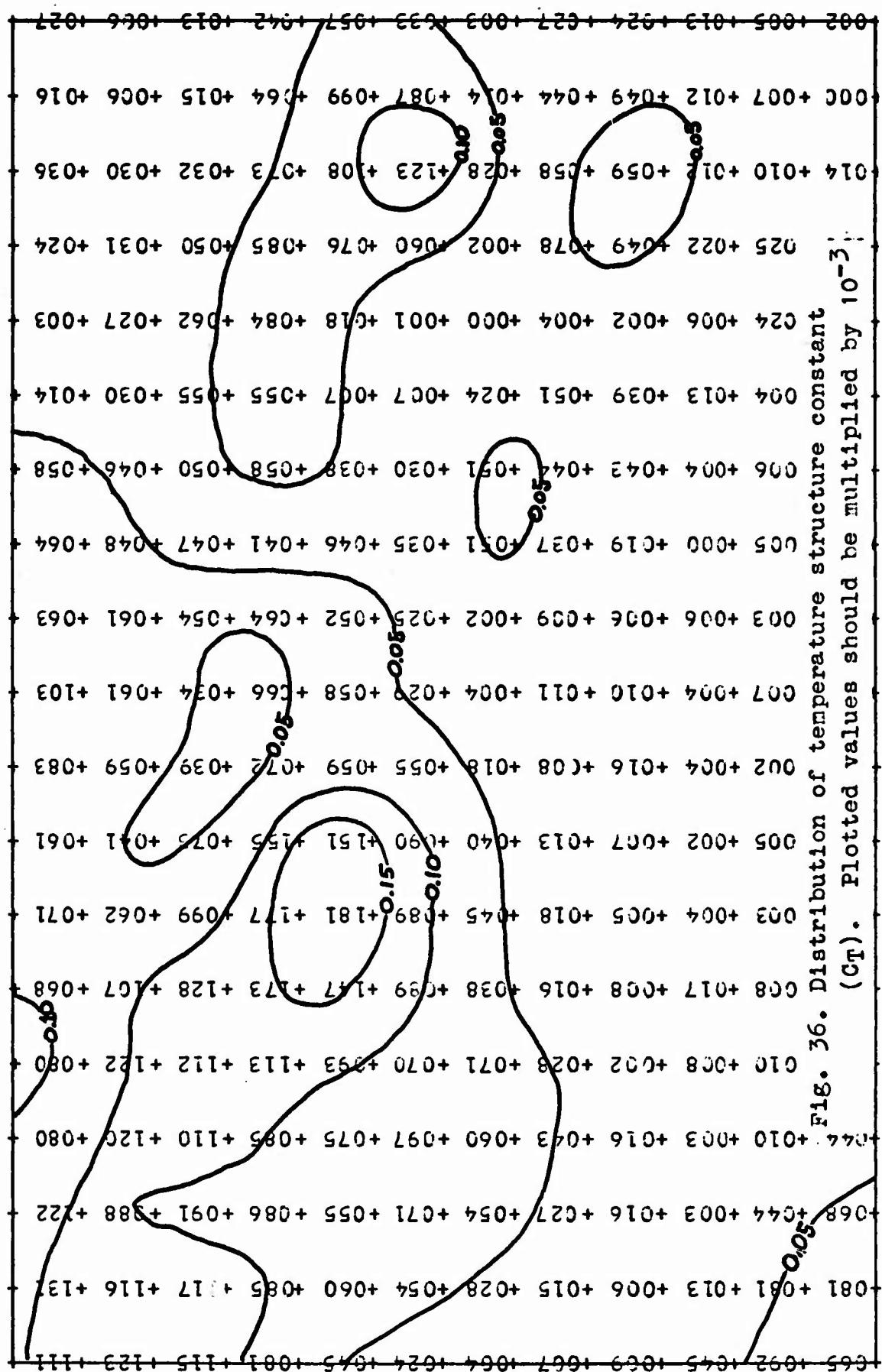
In general these results indicate that the two-layer model can produce surface wind fields which can be used for wave height analysis, even from a very simple wave model. It is noted that in the subtropics the wave heights computed from both sets of analysis are considerably below the observed wave heights. This is probably due to low surface geostrophic wind speeds in this area and the fact that no corrections are made for stability or baroclinicity at low wind speeds. This could possibly be corrected in the model by selecting a higher initial value for the drag coefficient.

#### E. RESULTS FROM INDEX OF REFRACTION COMPUTATIONS

A final application of the boundary layer model was the computation of the index of refraction structure constant ( $C_T^2$ ) using (84). The ratio of the exchange coefficients was determined from (85) and the input data for this equation were obtained by including the stability effect, the thermal wind effect, the correction for curvature, and using  $k = 0.35$ . All parameters needed to compute  $C_T^2$  were determined in the model.

The results from these computations appear in Figure 36. Values of  $C_T^2$  range from 0.0 in regions of near neutral stratification to a maximum of 0.181 in the region of maximum wind speeds. Comparisons of

Fig. 36. Distribution of temperature structure constant ( $C_T$ ). Plotted values should be multiplied by  $10^{-3}$ .



these values with those measured by Portman et al. (1968) indicate that they are comparable in regions of low wind speed with  $\theta_a - \theta_s$  less than 1 degree. Although no definite relation between  $C_T^2$  and wind speed or  $\theta_a - \theta_s$  appears in Figure 36, larger values of  $C_T^2$  appear to be associated with the greater wind speeds and larger instability ( $\theta_a - \theta_s$  ranging from -1 to -6°C).

The results indicate that a horizontal distribution of  $C_T^2$  for a given height, in this case 19.5 meters, can be computed with the model with some degree of confidence in the range of values obtained. Comparisons should be made between results computed by this model and observations of  $C_T^2$  to obtain better verification.

## VI. CONCLUSIONS

The results of this study have indicated that a marine boundary layer model can be applied to synoptic scale data to obtain various surface parameters. These parameters include friction velocity ( $U_*$ ), surface roughness ( $z_0$ ), surface Rossby number ( $R_o$ ), 19.5 meter wind ( $U_{19.5}$ ), veering angle ( $\Psi_o$ ), and stability length ( $z/L$ ).

First, the results indicate that the curvature correction produced the greatest change to the wind analysis. However, stability and modification of the von Kármán constant produced significant changes in the field, while changes due to the thermal wind were confined primarily to changes in wind direction.

Second, the 19.5 meter wind was computed including various physical specifications and corrections and compared with synoptic observations. The results indicate that the computed wind was lower than the observed wind in all areas and for all cases, but the general patterns agreed well with observations.

Finally, it appears that Cardone's planetary boundary model provides a reasonable framework to specify those surface layer parameters required for spectral wave forecasting and index of refraction calculations.

Further studies are needed to more completely verify and improve the model. The specification of the drag coefficient is probably the aspect in this model which needs the most attention. The drag coefficient

has received considerable examination in observational investigations and the results indicate some disagreement among investigators. A suggestion of this study is to increase the initial value of  $C_z$  to increase the computed surface wind speeds especially in areas of low wind speed. Application of this model to a region of reliable synoptic data such as during the BOMEX experiment in 1969 would reduce some of the errors due to data input and also test the physics of the model in the tropical atmosphere. Finally, observations of the index of refraction over a wide range of wind speed and stability would allow reliable comparisons to be made of the structure constant ( $C_T$ ).

PROGRAM TO COMPUTE A WIND AT 19.5 METERS USING BLACKADAR BAROCLINIC MODEL OVER A SPECIFIED GRID

PACIFIC GRID: INPUT 21 X 15, OUTPUT 19 X 13 - INPUT (1,1)  
CORRESPONDS TO FNWC (23,9) ON FORTRAN I,J GRID

DEFINITION OF PARAMETERS (UNITS OF MKS SYSTEM):

PRES = SURFACE PRESSURE (MB) FOR GRID POINTS  
TEMP = AIR TEMPERATURE (DEG C) FOR GRID POINTS  
STEMP = AIR SURFACE TEMP (DEG C) FOR GRID POINTS

DI = BLACKADAR CONSTANT  
C1, C2, C3, B = CONSTANT SPEED OF GEOGRAPHIC WIND

DI\*AH = ANGLES BETWEEN GEOSTROPHIC WIND AND THERMAL WIND

FI = CORTOLIS PARANTIER (/SEC) STABILITY PARAMETER

FI = GEOSTROPHIC WIND (MPS)

GAN = GEOSTROPHIC WIND ANGLE

PARAM = MODIFIED STABILITY LENGTH (L\*)

PHI = INFLUX ANGLE - COMPUTED FROM GAS EQUATION

RHOSS = DENSITY OF AIR - COMPUTED FROM GAS EQUATION

SD = N-N-DIMENSIONAL WIND SHEAR

TA = MEAN VERTICAL WIND ANGLE

TAN = N-N-DIMENSIONAL THERMAL WIND

TH = N-N-DIMENSIONAL THERMAL WIND

TORT = GEODISPID SPACING (M)

VST = FRICTION SPACING (U STAR) (MPS)

W195U = 19.5 METER WINDS (MPS)

W195U = 19.5 METER WIND (MPS)

XXK = YONKARMAN CONSTANT = 0.4

ZONL = ROUGHNESS PARAMETER (M)

ZONL = DIMENSIONLESS STABILITY PARAMETER (L STAR)

```

COMMON PRES(21,15),TEMP(21,15),D(21,15),S(21,15),W195U(21,15),W195V(21,15),
1)STEMP(21,15),W195(21,15),UGO(21,15),VGO(21,15)
1)READ(5,990)I1,J1
1)READ(5,1000)ENQ=9999,((PRES(I,J),I=1,J1),J=1,J1)
1)READ(5,1000)((TEND(I,J),I=1,J1),J=1,J1)
1)READ(5,1000)((STEMP(I,J),I=1,J1),J=1,J1)
DO 5 J=1,J1
DO 5 I=1,I1
5)IF(I,J)=TEMP(I,J) - STEM(I,J)
5)CALL WNDY63
999
STOP
1000
FORMAT(13F6.1)
END

```

SUBROUTINE WNDY63

THE FOLLOWING CONDITIONS MAY BE SET IN THE SUBROUTINE:

NEUTRAL CONDITION SET TD=0.0  
NO CURVATURE CORRECTION - SET CURVE = 0.0

CURVATURE CORRECTION - SET CURVE = 1.0  
NO THERMAL WIND CORRECTION - SET THERM = 0.0

COMPUTES GRIDPOINT PARAMETERS AND INITIAL 19.5 METER WIND COMPONENTS  
T2 IS DATE TIME GROUP (16 CHARACTERS)  
T3 IS DESCRIPTION OF RUN (48 CHARACTERS)

COMMON PRE(S(21,15),TEMP(21,15),DIF(21,15),S(21,15),W195(21,15),W195V(21,15)

1) STTEMP(21,15),W195(21,15),D(21,15),CT(21,15)

2) DIMENSION T2(4),T3(12),CT(21,15)

C1 = 6.84E-05

C2 = 4.28E-04

C3 = -4.43E-04

B = 3.0E-04

BB = 0.81

TA = 280.5

XK = 0.35

GRAV = 9.8

RDIAN = 3.14159/180.

READ(I,900) T2 T3

WRITE(6,1000) T2,T3

PR = 2.87E+06\*1.0E-04

DELTA = 3.81E-05

DO 400 J=2,14

400 J=2,20

J1 = J+8

RSQR = (J1-32)\*\*2 + (J1-32)\*\*2 + RSQR) / (973.752 + RSQR)

SINPHI = (973.752 - RSQR) / (973.752 + RSQR)

F = 2.0MEGA\*SINPHI

XW = 1.0E-0667 / (1.0E-0667 + SINPHI)

TDR T = 1.0 / (2.0\*DELT\*A\*XW)

SET F = CONSTANT FOR LE 15 DEG LAT

FTEST = SIN(15\*RADIAN)

IF(SINPHI\*LT\*FTEST) F = 2.0\*OMEGA\*FTEST

THRM = (GRAV\*TDRPT) / (F\*2.0\*(TEMP(I,J)+273.0))

RHO = PRFS(I,J) / (RR\*(TEMP(I,J)+273.0))

TD = DIF(I,J)

C

```

C COMPUTE GEOSTROPHIC WIND FROM SFC P FIELD
C
UGO(I,J) = -(TORT/(RHO*F))*(PRES(I,J+1) - PRES(I,J-1))
VGO(I,J) = -(TORT/(RHO*F))*(PRES(I+1,J) - PRES(I-1,J))
C V = SQRT(UGO(I,J)*2 + VGO(I,J)*2)
C V = 0.0
C IF(V < 0.0) GO TO 10
C IF(G.L = 0.0) GO TO 10
C CALL GRADWD(DELTA_XM,F,G,I,J)
G = SQR(TUGO(I,J)*2 + VGO(I,J)*2)
10 ZONL = 0.
RICH = 0.

C COMPUTE THERMAL WIND
C
TXO = THERM*(TEMP(I,J+1)-TEMP(I,J-1))
TYO = SORT(TXO*2 + TYO*2)
TH = SQRT(TXO*2 + TYO*2)
IF(TH.GT.100) TH = 100
CALL VANG(TUGO(I,J),VGO(I,J),GAN)
CALL VANG(TXO,I,TAN)
200 ETAH = TAN-GAN
210 IF(G.GT.5.0) GO TO 220

C FOR WIND SPEED LESS THAN 5.0 METERS/SEC, BYPASS BLACKADAR BAROCLINIC MODEL
C FOR CONSTANT DRAG COEFFICIENT = 0.0220, CONSTANT INFLOW ANGLE = 15 DEG
19.5 METER WIND = 0.7 GEOSTROPHIC WIND
C
C
C
VST = C*0.0220 * G
ZO = -C*3
PHI = 15.*RADIAN
W195(I,J) = 0.7*G
1.0*(G.L*0.0) GO TO 215
215 ZO = C1/VST + C2*VST*2 + C3
FE = 0.0
PARAM = 0.0
SD = 1.0
GO TO 340

C INITIALIZE BAROCLINIC MODEL
C
220 VST = 0.0245 * G
T = 0.0
ETA = 0.0
INDX = 1.0
IF(TD.GT.4.0) TD = 4.0
IF(TD.LT.-15.0) TD = -15.0
IF(ABS(TD).GT.1.0) GO TO 230

```

C ABSOLUTE VALUE OF AIR-SEA TEMPERATURE DIFFERENCE LE 1.0  
 C CONSTANT COEFFICIENT = 0.0245

ZONL = 0.0  
 PARAM = 0.0  
 FI = 0.0  
 FF = 0.0  
 SD = 1.0  
 GO TO 250

C ABSOLUTE VALUE OF AIR-SEA TEMP DIFFERENCE GT 1.0

230 PARAM = TAN(VST)\*\*2 / (1.0\*XX\*\*2\*GRAV\*TD)

240 ZONL = ((B\*G)/F)/PARAM  
 CALL PSI(ZONL,FI,RICH)  
 CALL SHR(ZCNL,SD,RICH)

C START BLACK-CAR BAROCLINIC MODEL

250 ZD = C1/VST + C2\*VST\*\*2 + C3  
 CPA = (B\*G/(F\*SD)) \* (ALOG(B\*G/(F\*ZD))-FI)

BET = SQRT(SD\*F\*\*2/(2.0\*VST\*XK\*B\*G))  
 F = SQRT(1.0 + 2.0\*CPA\*\*2\*BET\*\*2 + CPA\*BET)

R = SQRT(1.0 + (F\*\*2\*T\*\*2\*CPA\*\*2)/G\*\*2 - (2.0\*F\*T\*CPA\*COS(ETA))/G)

ALPH = ARCSIN(CPA\*BET/P)

ARGUM = (F\*T\*CPA\*SIN(ETA)/(R\*G))  
 GAMMA = ARCSIN(ARGUM)  
 PHI = ATAN(SIN(ALPH+GAMMA)/(P/R-COS(ALPH+GAMMA)))/(P\*SIN(PHI))

1IFI = ABS(S(VSTN-VST))

1IFT = ABS(VST - C.05)\*VST

260 VST = VSTN  
 GO TO 250

270 IF(ZCNL=0.0) 280,290,280

280 AA = 1.0/PARAM  
 CALL PSI(AA,FE,RICH)  
 ZD = C1/VST + C2\*VST\*\*2\*TAN(ALOG(1.0/ZD-FE))/XX\*\*2\*GRAV\*TD  
 PARAMNU = (VST\*\*2\*TAN(ALOG(1.0/ZD-FE))/XX\*\*2\*GRAV\*TD)  
 IF(ABS(PARAM-PARAMNU).LT.5.0) GO TO 250  
 PARAM = PARAMNU  
 GCT = 240

290 IF(INDEX-2) 300,310,300  
 300 IT = TH

ETA = ETAN  
 INDEX = 2  
 GO TO 250

```

310 IF (PARAM .NE. 0.0) GO TO 320
      ANEM = 0.0
      ZONL = 0.0
      GO TO 330
320 ZONL = 19.5/PARAM
      CALL PSI(ZONL,VST,ANEM,RICH)
      330 PHI1DFG(I,J) = PHI1/0.95(I,J)*COS(GAN+PHI)
      340 W195U(I,J) = W195(I,J)*SIN(GAN+PHI)
      W195V(I,J) = W195(I,J)*(SIN(GAN+PHI))
      IF(GOLEO*0.1) GO TO 342
      ROSS = G/(F*Z0)
      ROSSLG = ALOG10(ROSS)
      RUSTARG = VST/G
      GO TO 345
342 ROSSLG = 0.0
      RUSTARG = 99.99
      RATIO = 99.99
      C
      C CALCULATE INDEX OF REFRACTION (CT)
      C
      345 ZHT = 19.5
      IF (PARAM .NE. 0.0) GO TO 350
      ZL = 0.0
      XKHKM = 1.0
      GO TO 360
350 ZL = ZHT/PARAM
      IF (ZL .LT. 0.06) GO TO 355
      XKHKM = 1.0
      GO TO 360
355 XKHKM = 1.35*(SQR((1.0-ZL)/(1.0-FE)) 15.*ZL)**0.25)
      360 DTDZ = (SD*TD)/(ZHT)
      CT11 = (XK*ZHT)**1.3333333
      CT22 = (SD-ZL)**2.3333333
      CT(I,J) = SQR((BB*XKHKM*CT11*DTDZ**2)/(SD*CT22))
      WRITARG*XKHKM,CT(I,J),RICH
      1 CONTINUE
      400 RETURN
      900 FORMAT(4A4/12A4)
      1000 1. J*2X*AIR-SEA DIFF,2X*GEOST WIND,2X*ROUGH PARAM,3X*Z/L*3
      2 X*PHI(DEG),2X*USTAR,2X*W195,2X*LOG ROSSBY NO.,2X*USTAR/G,
      3 2 X*ALPHA,2X*UDFR,2X*RI,2X*RF,4X*RI,2X*RI
      110C 1 FORMAT(1X*12*2X*12*5XF5.1*8XF4.1*6XF8.6,2XF7.3,1,4XF5.3
      1 2 X,F4.1,7X,F4.2,6X,F6.4,4XF4.2,4XF5.3
      END

```

```

C SUBROUTINE PSI (P,B,C)
C IF (P.GT.0.) GO TO 10
C CALL SHR (P,S,D)
C B = 1.-S.-3.* ALOG (S)+2.* ALOG ((1.+S)/2)+2.* ATAN (S)-1.5708+AL06((1.
C 1+S.*#2)/2.)
C C = D
C RETURN
C 10 B=-7.*P
C C = P
C RETURN
C END

C SUBROUTINE SHR (PS,B,C)
C IF (PS.GT.0.) GO TO 30
C RI=PS
C RINEW=PS*(1.0-18.*RI)**(1.-G0/4.)
C 10 IF (ABS (RINEW-RI).LT.0.01) GO TO 20
C RI=RINEW
C GO TO 10
C B=1. / (1.-18.*RINEW)*(1. / 4.)
C 20 C=RINEW
C RETURN
C 30 B=1. +7.*PS
C C = PS
C RETURN
C END

C SUBROUTINE GRIDEQ (I,J,DEGLAT,DEGLON)
C SUBROUTINE TO COMPUTE LAT-LONG VALUES FROM FNWC I FROM GRID
C LONGITUDE IN DEGREES AND DECIMAL FRACTIONS, + WEST FROM 0 DEGREES.
C
C RE = 31.205**2
C YP = 1 - 32.
C XP = J - 32.
C DEGLAT = 57.2957795*ARSIN((RE-XP**2-YP**2)/(RE+XP**2+YP**2))
C IF (XP.NE.0.) GO TO 20
C IF (YP.NE.0.) GO TO 103,104,105
C 102 ANG = C#
C 103 GO TO 50
C 104 ANG = 80.
C GO TO 50.

```

```

105 ANG = 260.
    GO TO 50
20 IF(ANG<=350.0) ANG = 350.0 - ATAN2(YP,XP)*57.2957795
    ANG = 360.0 + ANG
    GO TO 50
50 IF((ANG - 360.0).LE.0.0) GO TO 55
    ANG = ANG - 360.
55 DEGLON = ANG
    RETURN
    END

```

```

C SUBROUTINE GEO GV
C CALCULATE GEOGRAPHIC WIND DIR(D) AND SPEED(S) FROM U AND V
C COMMON PRESETS(21,15),TEMP(21,15),DIF(21,15),S(21,15),W195U(21,15),
1) S(21,15),W195(21,15),D(21,15),UGO(21,15),VGO(21,15)
P1 = 3.14159
DO 10 J=2,14
    DO 10 I=2,2C
        K=I+2
        CALL GRICEG(K,L,ALAT,ALON)
        AA=W195U(I,J)
        BB=W195V(I,J)
        CALL VANG(AA,BB,ANGLE)
        ANGLE = ANGL1+180./3*14159
        D(I,J) = ALON + ANGLE - 260.
        IF(D(I,J).LT.0.5) D(I,J) = 360. + D(I,J)
        D(I,J) = ANGDD(D(I,J),360.)
        S(I,J) = SQRT(W195U(I,J)**2 + W195V(I,J)**2)
10    CONTINUE
    RETURN
    END

```

```

C SUBROUTINE VANG(A,B,ANGLE)
C COMPUTE ANGLE IN I,J COORDINATES: A,B ARE I,J COMPONENTS RESPECTIVELY,
C ANGLE IS COMPUTED ANGLE IN RADIAN
    IF(A.NE.0.0 AND B.NE.0.0) GO TO 20
    IF(A.NE.0.0) ANGLE = 90.
    IF(B.LT.0.0) ANGLE = 270.
    IF(A.EQ.0.0 AND B.EQ.0.0) ANGLE = 0.0

```

```

10 GO TO 30
10 ANGLE = 0.0 ANGLE = 180.
GO TO 30
20 ANGLE = ATAN2(B,A)*180./3.14159
10 ANGLE = LT*0.0 ANGLE = 3.60.+ ANGLE
30 ANGLE = ANGLE*3.14159/180.
RETURN
END

```

```

C SUBROUTINE GRADWD(A,B,C,E,I,J)
C SUBROUTINE TO COMPUTE A GRADIENT WIND USING CURVATURE AND THE GEOSTROPHIC
C WIND
COMMON PRES(21,15),W195(21,15),D15(21,15),D195(21,15),WGO(21,15)
11 FMAX = 0.75
FMIN = -0.2/16.
DLTAM = A/B
XCSIN = UGO(I,J)/E
XCSIN = YGO(I,J)/E
DPDX = (PRSS(I+1,J) - PRES(I-1,J))/(2.0*DELTAM)
DPDY = (PRSS(I,J+1) - PRES(I,J-1))/(2.0*DELTAM)
D2PDX = (PRSS(I+1,J+1) - 2.*PRSS(I,J) + PRES(I-1,J))/(DELTAM**2)
D2PDY = (PRSS(I,J+1) - 2.*PRSS(I-1,J+1) + PRES(I-1,J+1) - PRES(I+1,J-1) + PRES(I-1,J-1))/(DELTAM**2)
1 IF(DPDY/(4.0*(DLTAM**2)) < J-1) GOTO 10
5 ANUM = (DPDY+2)*SQR((1.0 + (DPDX/DPDY)**2)**3)
6 DENOM = DPDX*(2.0*D2PDX - D2PDY*(DPDX/DPDY)) - DPDX*D2PDY
1 IF(DENOM .EQ. 0.0) GOTO 10
1 IF(ANUM .EQ. 0.0) GOTO 10
RCURV = ANUM/DENOM
YGRF = E/(RCURV)
YGRF = E/(RCURV)
1 IF(YGRF .LT. FMIN) YGRF = FMIN
1 IF(YGRF .GT. FMAX) YGRF = FMAX
ROOT = 1/(0.5 + SQR(0.25+YGRF))
E = *ROOT
UGO(I,J) = XCOS**E
VGO(I,J) = XSIN**E
10 RETURN
END

```

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